DEPARTMENT OF THE ARMY TECHNICAL MANUAL

DIRECT AND GENERAL SUPPORT MAINTENANCE MANUAL

ARMAMENT SUBSYSTEM, HELICOPTER, 7.62 MILLIMETER MACHINE GUN: HIGH RATE, XM27E1 (1005-933-6242) (USED ON OH-6A HELICOPTERS)

This copy is a reprint which includes current pages from Changes 1 through 3

HEADQUARTERS, DEPARTMENT OF THE ARMY JULY 1969

WARNING

Warning: Do not attempt to troubleshoot a loaded weapon. Be sure no ammunition is present in any subsystem component before energizing the system.

Warning: Clear the weapon before starting inspection. Point weapon in a safe direction and determine if live rounds are present. Check bore and chamber for obstructions; for example, a bullet in the bore or a ruptured cartridge case in the chamber.

Warning: Personnel operating vapor degreaser are cautioned not to breathe the vapor fume.

Warning: Be sure there is no ammunition in any subsystem component before operating the electric gun drive.

Warning: If a power sander is used, avoid breathing the fine fiberglass dust produced. This dust can cause serious skin and respiratory irritations. Wear protective clothing and use a mask.

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, D. C. 8 October 1970

DIRECT AND GENERAL SUPPORT MAINTENANCE MANUAL

ARMAMENT SUBSYSTEM, HELICOPTER,

7.62 MILLIMETER MACHINE GUN:

HIGH RATE, XM27E1

(1005-933-6242)

(USED ON OH-6A AND OH-58A HELICOPTERS)

TM 9-1005-298-34, 30 July 1969, is changed as follows:

Change the title to read: DIRECT AND GENERAL SUPPORT MAINTENANCE MANUAL

ARMAMENT SUBSYSTEM, HELICOPTER,

7.62 MILLIMETER MACHINE GUN:

HIGH RATE, XM27E1

(1005-933-6242)

(USED ON OH-6A AND OH-58A HELICOPTERS)

1

Change No. 1

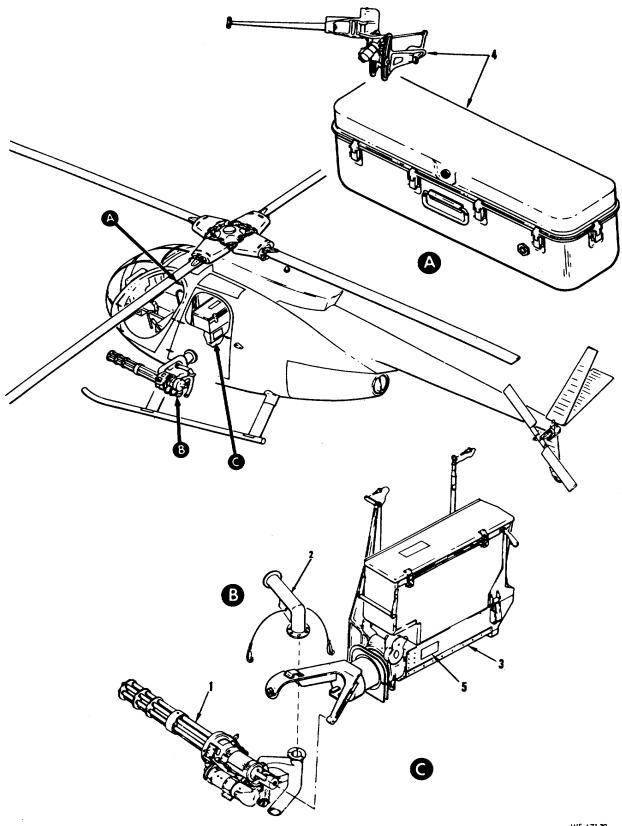
Page i:

*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*
*																			
СН	PTEF	r IV. I	REPAI	R OF	RAM	<u>AIR D</u>	<u>UCT</u> 4	SSEN	MBLY										
*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*
*																			

Page 1-2:

Delete figure 1-1, WE 15616A, and substitute figure 1-1, WE 67170, Components of Armament Subsystem XM27E1.

2



WE 671.70

Figure 1-1. (Superseded) Components of Armament Subsystem.

Page 3-5:

Delete figure 3-3, WE 61488, and substitute figure 3-3, WE 67172, Bolt Assembly-exploded view.

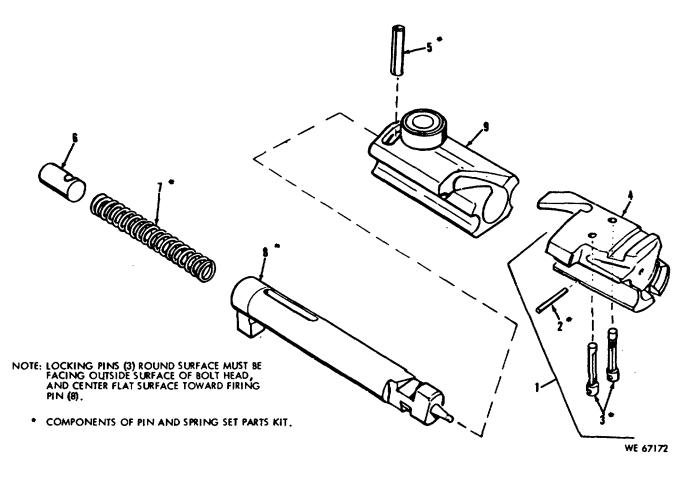


Figure 3-3. (Superseded) Bolt Assembly-exploded view.

Page 4-1:

CHAPTER 4 (SUPERSEDED) REPAIR OF RAM AIR DUCT ASSEMBLY

4-1. Description

The ram air duct assembly is a metal air scoop, consisting of the ram air duct, three turnlock fasteners, retaining cable, two snap fasteners, and a rubber grommet.

4-2. Disassembly/ Assembly

See figure 4-1. Limit disassembly to removal of faulty parts such as fasteners, retaining cable, and rubber grommet.

4-3. Cleaning, Inspection, and Repair

a. Cleaning. Refer to paragraph 2-9a and apply procedures used for cleaning mount assembly parts.

b. Inspection. Check for damaged fasteners, frayed or broken retaining cable, and for torn or miming rubber grommet.

c. Repair. Replace excessively worn, broken, or miming parts using standard shop procedures and tools.

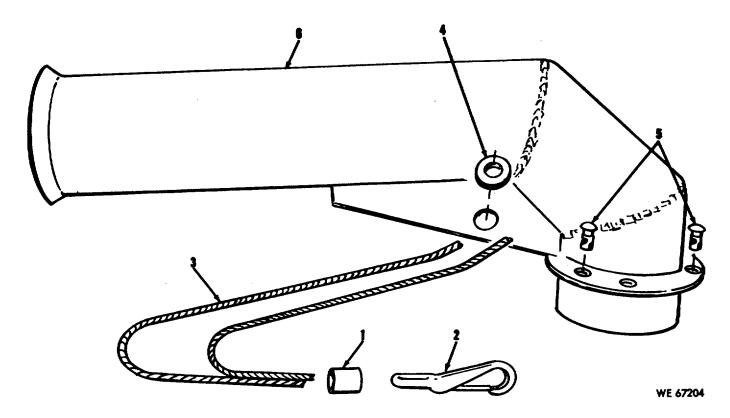


Figure 4-1. Ram air duct assembly-exploded view.

Page 5-16:

5-9.	Clea	ning,	Inspe	ction,	and R	lepair													
*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*
*																			
<i>b</i> . I	nspec	tion																	
*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*	*
*																			

(4) (Superseded) Check tube assembly parts including resilient mount (11, fig. 5-8), bolt ball (2), and camlock (9) for excessive wear or damage.

Delete figure 5-8, WE 15607, and substitute figure 5-8 WE 67241, tube assembly-partially exploded view.

Page 5-20:

5

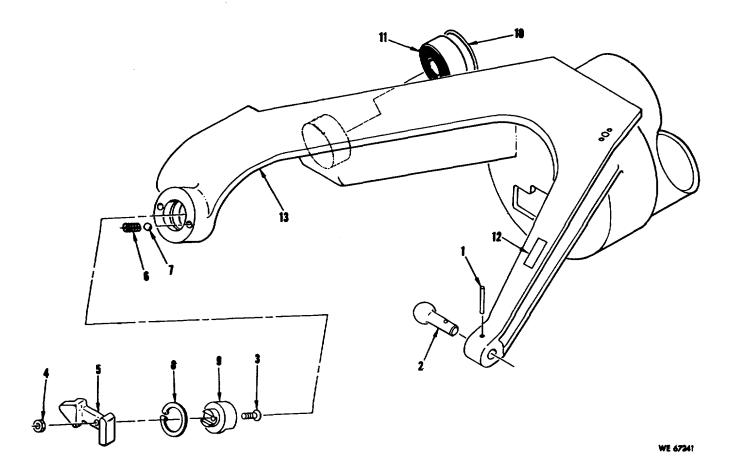


Figure 5-8 (Superseded) Tube Assembly-partially exploded view.

Delete the key for figure 5-8, WE 15607, and substitute the following key for figure 5-8, WE 67241:

- 1. Spring Pin
- 2. Bolt Bail
- 3. Screw
- 4. Nut
- 5. Knob, Lock
- 6. Spring
- 7. Ball

- 8. Retaining Ring
- 9. Cam Lock
- 10. Retaining Ring
- 11. Mount, Resilient
- 12. Decal
- 13. Support Assembly

Page 6-8:

Delete figure 6-4, WE 15615, and substitute figure 6-4 WE 52857, Reticle project assembly-partially exploded view.



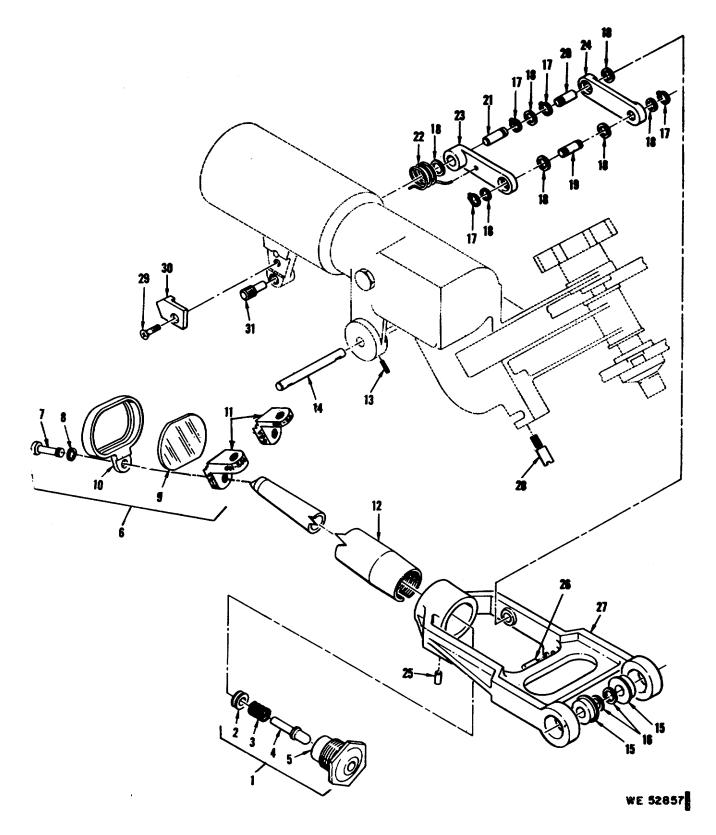


Figure 6-4 (Superseded) Reticle project assembly—partially exploded view.

Page 6-9:

Delete key for figure 6-4, and substitute the following:

- Detent Assembly
 Spring Retainer
 Spring
 Plunger
 Body
 Beamsplitter Assembly
 Screw
 Washer
 Lens
 Retainer
 Adapter
 Arm
 Setscrew
 Shaft
 Bearing
- 16. Washer

By Order of the Secretary of the Army:

Official:

KENNETH G. WICKHAM, Major General, United States Army, The Adjutant General.

Distribution

To be distributed in accordance with DA Form 12-31 (qty rqr block no. 129) Direct and General Support Maintenance requirements for Armament Subsystem XM27E1.

8

- Retainer Ring
 Washer
 Shaft
 Shaft
 Shaft
 Shaft
 Shaft
 Shring
 Arm
 Arm
 Arm
 Spring Pin
 Spring Pin
 Arm
 Screw
 Screw
 Screw
 Screw
 Bracket
- 31. Eccentric Pin

W. C. WESTMORELAND, General, United States Army, Chief of Staff. Changes in force: C 1, and C 2

Change

No. 2

HEADQUARTERS, DEPARTMENT OF THE ARMY Washington, D.C., 23 September 1971

Direct and General Support Maintenance Manual ARMAMENT SUBSYSTEM, HELICOPTER, 7.62 MILLIMETER MACHINE GUN: HIGH RATE XM27E1 (1005-933-6242) (USED ON OH-6A AND OH-58A HELICOPTERS)

TM 9-1005-298-34, 30 July 1969, is changed as follows: *Page 2-11*. Paragraph 2-7e is superseded as follows:

e. Threads/Screw Thread Inserts.

(1) Repair of damaged threads shall be accomplished by use of a thread restorer, by chasing on a lathe, or by use of a tap of proper dimension.

(2) All screw thread inserts shall be installed in accordance with MS 33646. Remove notched tangs after installation. Damaged or defective inserts will be replaced. If hole is enlarged during removal of insert, an oversized insert should be used as replacement.

Page 3-5. Paragraph 3-5 is superseded as follows:

3-5. Disassembly/Assembly.

a. Disassembly. Refer to the numerical sequence in figures 3-4 and 3-4.1 and the following specific instructions for disassembly of the feeder and feeder housing assembly.

(1) Drive out pin (14, fig. 3-4) while plate (18) is still in place to support shaft (31).

(2) If pin (26) is badly bent and difficult to drive out, the following method of removal may be used.

(a) Position gear (27) so that it rests on a wooden block(s), with clearance for shaft (31).

(b) Strike shaft (31) sharply with soft-faced hammer to shear pin (26).

(c) Drive out pieces of pin (26) from gear (27), push rod guide (30), and shaft (31).

(3) When removing spring-loaded solenoid (45) assure that all matching parts are kept together for installation as a unit.

b. Assembly. Assembly of the feeder and feeder housing assembly is in reverse order of disassembly and the following instructions.

(1) After installation, face of identification plate (50, fig. 3-4) shall be coated with Lacquet Spec TT-L-58,

Type 1, Clear.

WARNING

Solenoid (45) consists of matched parts. Use only as a unit. Interchange of solenoid parts may result in a firing system malfunction.

(2) Install solenoid (45) using only matched parts.

(3) Install pin (29) with approximately 1/8 inch protruding.

NOTE

Pin (29) is for alinement of recess in inside diameter of gear (27) over push rod guide (30).

(4) Use 3/16 inch punch and aline holes in gear (27), push rod guide (30), and shaft (31) so that a slot in guide is opposite point where punch enters gear.

(5) Hold gear (27) - push rod guide (30) - shaft (31) unit firmly, remove punch and install pin (26) through unit. Pin shall not protrude into slot of guide nor restrict gear tooth action.

NOTE

Stripper (23) is installed on shaft (31) so that slot in line with hole is in alinement with slot (para (4)) in push rod guide (30) having hole for pin (26).

(6) Install stripper (23) to shaft (31) with pin (22) entering from slot alined hole in stripper. Pin shall be flush with surface of stripper.

(7) Press bearing (19) into guide plate (20) until flange of bearing is flush with surface of plate.

(8) Slide sprocket (21) onto shaft (31) so that alinement pin (24) of stripper (23) engages recess in

sprocket sleeve and install plate (18).

(9) Install pin (16) in sprocket (21) so that pin is flush with surface of sleeve.

NOTE

Prior to installation of instruction plate (9, fig. 3-4.1) adhesive on back shall be activated by use of acetone, methyl-ethyl-ketone, or methylisobutyl-ketone.

(10) Clean surface to receive instruction plate (9, fig. 3-4.1) so that it is free of oil, dirt, grease or other contaminants that could impair adhesion. Activate adhesive on back of plate and install on housing (12) using firm overall pressure.

(11) Install pins (8) so that approximately 3/8 inch protrudes on side toward instruction plate (9).

(12) After complete assembly of feeder perform the following checks and tests.

(a) Rotate sprocket (21, fig. 3-4) and shaft (31) to check for smooth operation.

(b) Check sprocket (21) and gear (27) to assure that there is no end play on shaft (31).

(c) Slide a round of ammunition through each opening (slot) in sprocket (21) and observe that there is no catching or binding.

(d) Check that clearing solenoid (45) is operating smoothly.1

NOTE

Nose guide 11686381 (7, fig. 3-4), issued as part of a feeder, is replaced for XM27E1 only at assembly by nose guide 11697451, issued as part of gun assembly. Refer to TM 9-1005-298-12.

Pages 3-6, 3-7, and 3-8. Figure 3-4 is superseded as follows:

2

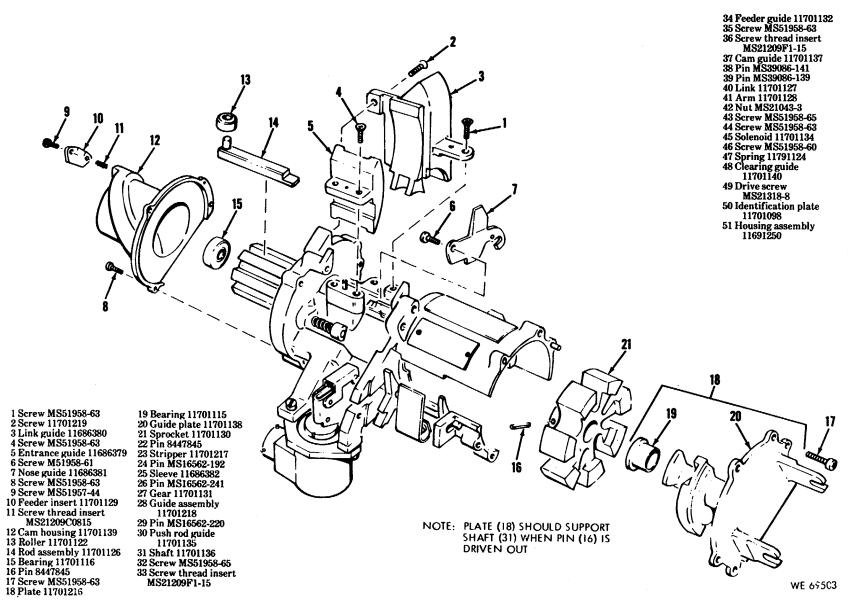


Figure 3-4. Delinking feeder MAU-56/A-exploded view (sheet 1 of 2)

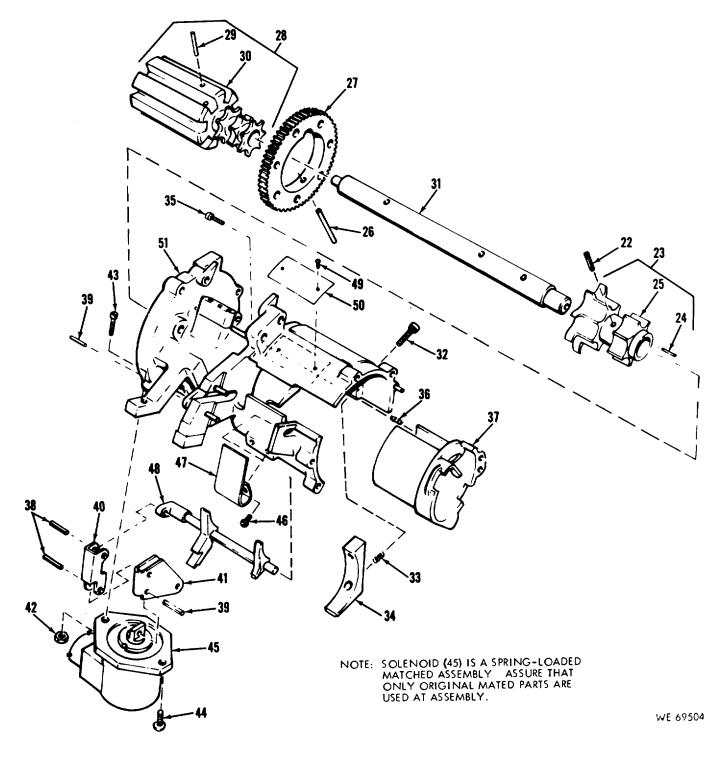


Figure 3-4. Delinking feeder MAU/56-A-exploded view (sheet 2 of 2)

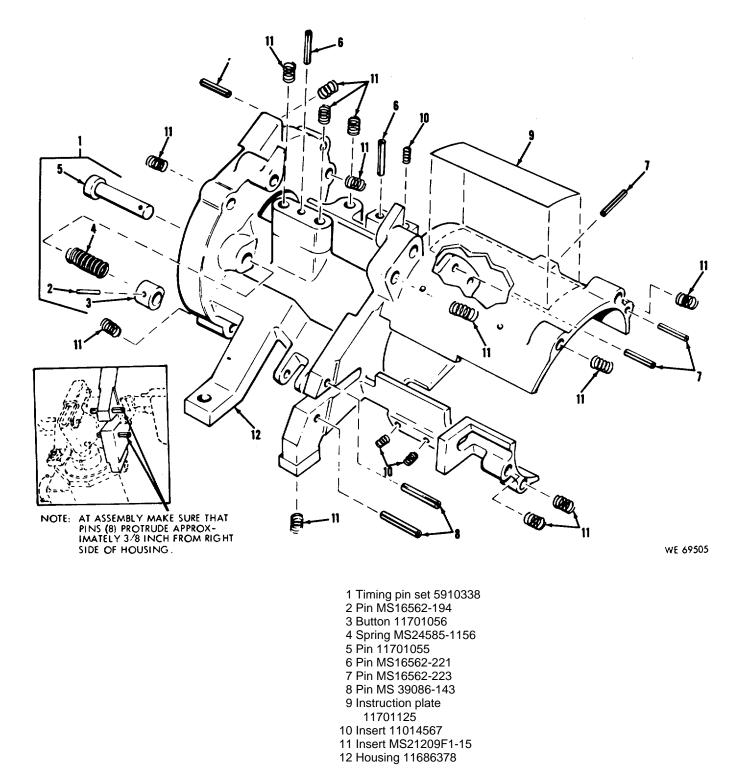


Figure 3-4.1. Feeder housing assembly—exploded view.

W. C. WESTMORELAND, General, United States Army,

Chief of Staff

By the Order of the Secretary of the Army:

Official:

VERNE L.. BOWERS, Major General, United States Army, The Adjutant General.

Distribution:

To be distributed in accordance with DA Form 12-31, (qty rqr block no. 129) Direct and General Support maintenance requirements for 7.62MM Machine Gun Armament Subsystem XM27E1.

6

Changes in force C 1, C 2, and C 3

HEADQUARTERS DEPARTMENT OF THE ARMY WASHINGTON, DC, 17 December 1975

Aviation Intermediate Maintenance Manual

ARMAMENT SUBSYSTEM, HELICOPTER 7.62 MILLIMETER MACHINE GUN: HIGH RATE M27E1 (NSN 1005-00-933-6242) (USED ON OH-6A HELICOPTERS)

TM 9-1005-298-34 30 July 1969, is changed as follows:

The title is changed as shown above.

Page i. Immediately following the title add the following:

To implement the three level maintenance concept, the following changes will be made to this manual, as applicable:

a. Substitute the words Aviation Unit maintenance for Crew/Operator and Organizational maintenance rst level of maintenance). Also, wherever the symbol for Crew/Operator maintenance (C) is used, change to .the symbol for Aviation Unit maintenance (O).

b. Substitute the words Aviation Intermediate maintenance for Direct Support and General Support maintenance (second level of maintenance). Also, wherever the symbol for General Support maintenance (H) is used, change to the symbol for Aviation Intermediate maintenance (F).

c. The Depot level of maintenance remains the same (third level of maintenance).

d. Under the new three level maintenance concept, the maintenance codes are as follows: Aviation Unit Maintenance (O), Aviation Intermediate Maintenance (F), and Depot Maintenance (D).

By Order of the Secretary of the Army:

Official:

FRED C. WEYAND General, United States Army Chief of Staff

PAUL T. SMITH Major General, United States Army The Adjutant General

Distribution:

To be distributed in accordance with DA form 12-31 (qty rqr block No. 412), Direct/General Support requirements for 7.62-MM Machine Gun, High Rate, M27.

CHANGE NO. 3 TECHNICAL MANUAL

}

No. 9-1005-98-34

HEADQUARTERS, DEPARTMENT OF THE ARMY WASHINGTON, D. C., *30 July 1969*

DIRECT AND GENERAL SUPPORT MAINTENANCE MANUAL

ARMAMENT SUBSYSTEM, HELICOPTER, 7.62 MILLIMETER MACHINE GUN: HIGH RATE, XM27E1 (1005-933-6242) (USED ON OH-6A HELICOPTER)

This manual is current as of 24 February 1969.

			Paragraphs	Pages
CHAPTER	1.	INTRODUCTION		
SECTION	Ι.	General	1-1, 1-2	1-1
SECTION	II.	Description and data	1-3, 1-4	1-3
CHAPTER	2.	DIRECT AND GENERAL SUPPORT		
		MAINTENANCE INSTRUCTIONS		
SECTION	Ι.	Repair parts, special tools, and equipment	2-1, 2-2	2-1
SECTION	II.	Troubleshooting	2-3	2-1
SECTION	III.	Preembarkation inspection of material in units alerted for		
		overseas movement	2-4, 2-5	2-10
SECTION	IV.	General maintenance	2-6, 2-9	2-10
SECTION	V.	Removal and installation of major components	2-10	2-14
CHAPTER	3.	REPAIR OF GUN ASSEMBLY		
SECTION	Ι.	Gun assembly	3-1	3-1
SECTION	II.	7.62 Millimeter gun M134	3-2, 3-3	3-1
SECTION	III.	Delinking feeder MAU-56/A	3-4, 3-6	3-5, 3-9
SECTION	IV.	Electric gun drive assembly	3-7, 3-11	3-9, 3-10
SECTION	V.		3-12, 3-13	3-13
CHAPTER	4.	REPAIR OF FAIRING ASSEMBLY	4-1, 4-3	4-1
CHAPTER	5.	REPAIR OF MOUNT ASSEMBLY		
SECTION	Ι.	Mount assembly	5-1, 5-3	5-1
SECTION	II.	Ammunition container assembly	5-4, 5-5	5-3
SECTION	III.	Electrical system assembly	5-6, 5-7	5-5
SECTION	IV.	Housing and tube assembly	5-8, 5-10	5-16
CHAPTER	6.	REPAIR OF HELICOPTER REFLEX SIGHT XM70E1	6-1, 6-6	6-1
CHAPTER	7.	FINAL INSPECTION	7-1	7-1
APPENDIX	Α.	REFERENCES		A-1

*This manual supersedes TM 9-1005-298-35, 19 May 1967.

CHAPTER 1

INTRODUCTION

Section I. GENERAL

1-1. Scope

a. This manual contains instructions for direct and general support maintenance of armament subsystem XM27E1 (fig. 1-1).

b. The maintenance allocation chart in TM 9-1004-28-12 allocates maintenance responsibilities.

1-2. Forms and Records

a. Authorized Forms. DA Forms and procedures used for equipment maintenance will be only those prescribed in TM 38750, Procedures. Refer to DA Pam

310-2 for a listing of all forms.

b. Recommendations for Maintenance Manual Improvements. Report of errors, omissions, and recommendations for improving this publication 'by the individual user is encouraged. Reports should be submitted on DA Form 2028 (Recommended Changes to DA Publications) and forwarded direct to: Commanding General, U. S. Army Weapons Command, ATTN: AMSWE-SMM-P, Rock Island, Illinois 61201.

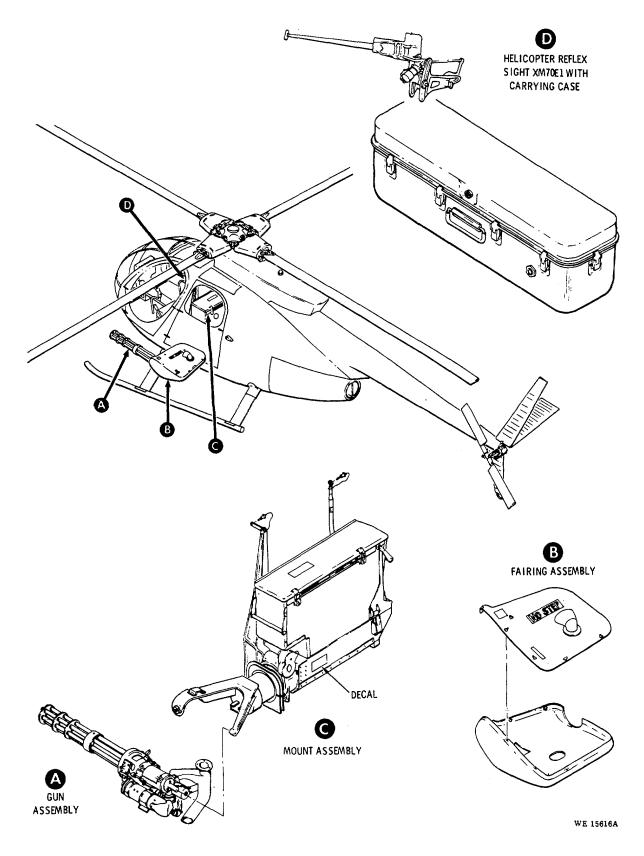


Figure 1-1. Components of the armament subsystem XM27E1.

Section II. DESCRIPTION AND DATA

1-3. Description Refer to TM 9-1005-298-12.

1-4. Tabulated Data

Refer to TM 9-1005-298-12.

CHAPTER 2 DIRECT AND GENERAL SUPPORT MAINTENANCE INSTRUCTIONS

Section I. REPAIR PARTS, SPECIAL TOOLS AND EOUIPMENT

2-1. Special Tools and Equipment

Special tools and equipment are listed in TM 9-1 005-298-35P and in table 2-1.

2-2. Direct and General Support Maintenance Repair Parts

Direct and general support maintenance repair parts are listed and illustrated in TM 9-1005-298-35P.

Section II. TROUBLESHOOTING

2-3. General

Troubleshooting is presented in two tables. Table 2-2 provides troubleshooting for defective mechanical components and table 2-3 provides troubleshooting for the electrical system. The tables include tests and corrective actions applicable to direct support and general support levels of maintenance and should be

considered an extension of the organizational level troubleshooting tables to be found in TM 9-1005-298-12

a. When performing electrical troubleshooting, test equipment will be required as noted in the procedures. The schematic diagram, figure 2-1 and wiring diagram, figure 2-2, should be used as an aid in tracing the circuits.

		R	eference	
Identify	ing			
Item	Number	Fig.	Para.	Name and use
1	111697520 (composed of 11697521, 11697522, and 11697523)	5-11	5-10	Too combination for adjusting the forward bearing housing assembly.
2	182350		6-4	Dioptometer, Keuffel and Esser No. 182350 (or equivalent), to check focus of sight projector reticle.
3	10547466 (4931-936-54031	6-9	6-6	Fixture for final inspection and colli- mation of sight X-M70E1.

b. Connect external 28 VDC electrical power to the aircraft to prevent excessive battery drain.

c. When a troubleshooting check calls for energizing the system, the armament panel and aircraft controls will be positioned as follows:

Warning: Do not attempt to troubleshoot a loaded weapon. Be sure no ammunition is present in any subsystem component before energizing the system.

(1) BATT-OFF-EXT switch at EXT (or BATT if external power is not available).

(2) ARM and ARM POWER circuit breakers depressed.

(3) MODE MASTER switch at FIRE NORM.

(4) ARM-SAFE switch at ARM.

(5) Normal indications at this time will be GUN NOT CLEARED light off, ARMED light on, and AMMO LOW light on.

Note. A preliminary test of all indicator lights on the armament panel can be made by depressing the LIGHTS TEST pushbutton at the upper center of the air-craft instrument panel electrical equipment repairman.

d. When electrical troubles are isolated to aircraft wiring or component, notify the aircraft electrical equipment repairman.

Malfunction	Probable cause	Corrective action
	Gun Assembly	
Gun fails to rotate	Defective drive motor.	Replace driver motor or repair as required.
Damaged delinking feeder.	Replace or repair as required.	
Gun stops firing	Damaged rotor assembly.	Repair rotor assembly.
Gun fails to feed	Damaged or broken guide bar.	Replace guide bar.
	Bent or broken fingers on gun housing.	Repair or replace housing.
	Damaged or broken extractor on bolt head.	Replace firing bolt head assembly.
Gun fails to extract	Bent or broken guide bar allows round to	Replace guide bar.
	to feed ahead of bolt assembly.	
	Damaged or broken extractor on bolt head.	Replace firing bolt assembly.
	Damaged gun housing.	Replace gun housing.
	Mount Assembly	
Mount will not elevate or depress	Defective clutch or drive gearing.	Repair or replace as required.
	Defective gear sector or worm gear.	Repair or replace torque tube assembly.
	Torque tube binds in housing.	Replace torque tube assembly.
Ammunition does not feed properly.	Defective feed chute or elbow chute.	Repair or replace as required. v
	Worn or defective rollers or levelers in	Repair or replace as required.
	ammunition container.	
	Helicopter Reflex Sight XM70E1	
Reticle pattern not centered on beam- splitter.	Bent or damaged beamsplitter arm.	Replace arm assembly or beamsplitter.
•	Locking detent worn or damaged.	Repair or replace.
Reticle pattern canted from vertical and horizontal.	Reticle not properly positioned in projector.	Adjust reticle or projector assembly.
Reticle pattern not clear and sharp.	Optical elements not positioned for proper focus.	Adjust as required or replace projector.
	Defective rheostat.	Replace rheostat.
	Moisture inside projector due to failure of	Repair as required or replace projector.
	hermetic sealing.	
Elevation control knob binding.	Improper lubrication or contamination.	Disassemble, clean, and lubricate.
ő	Bent shaft.	Replace.
Looseness between fitted parts of sight.	Excessive wear.	Repair or replace as required.
. 0	Improper tightening of attaching parts.	Tighten as required.
	1	

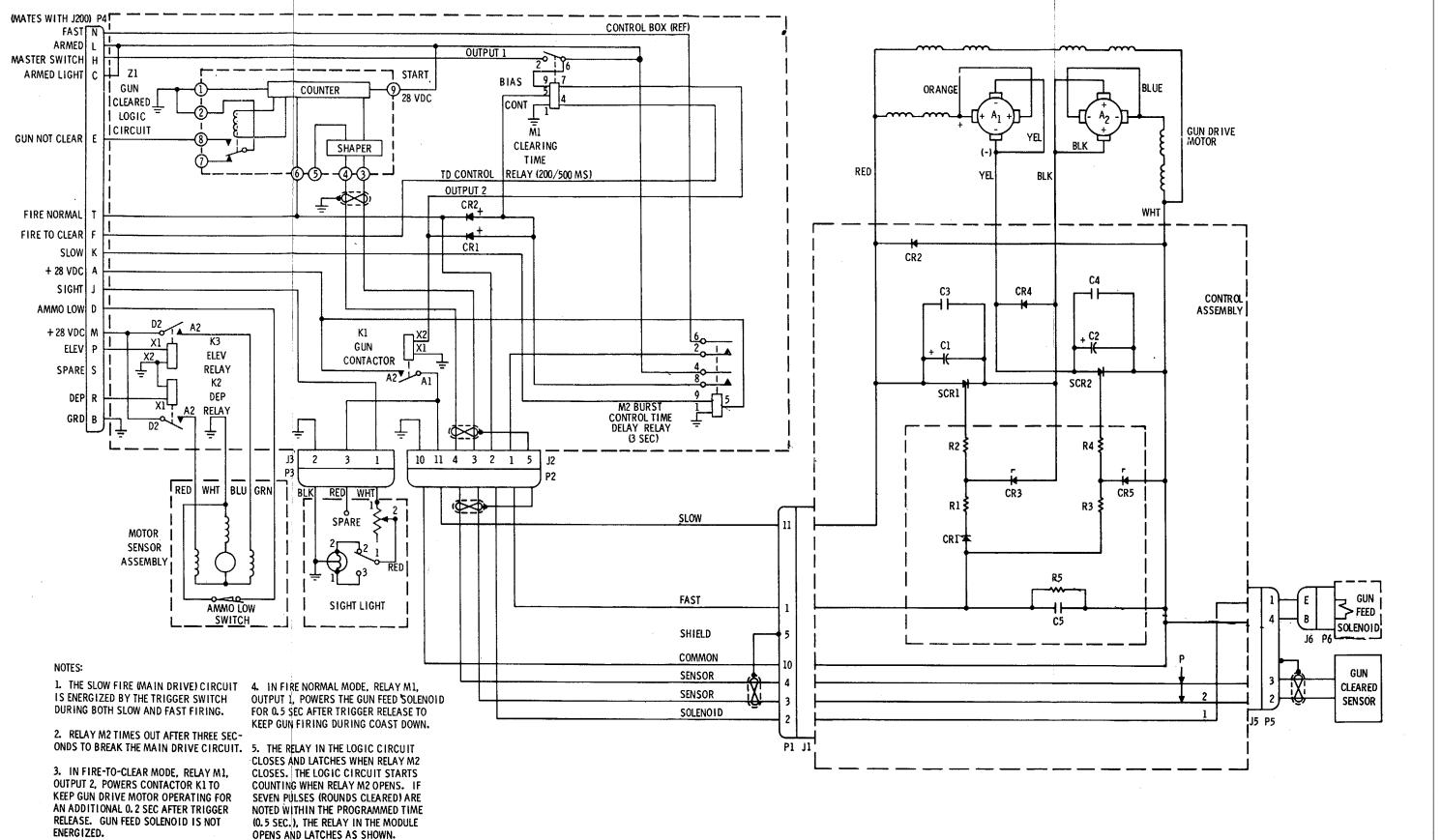


Figure 2-1. Schematic diagram-armament subsystem XM27E1.

WE61285

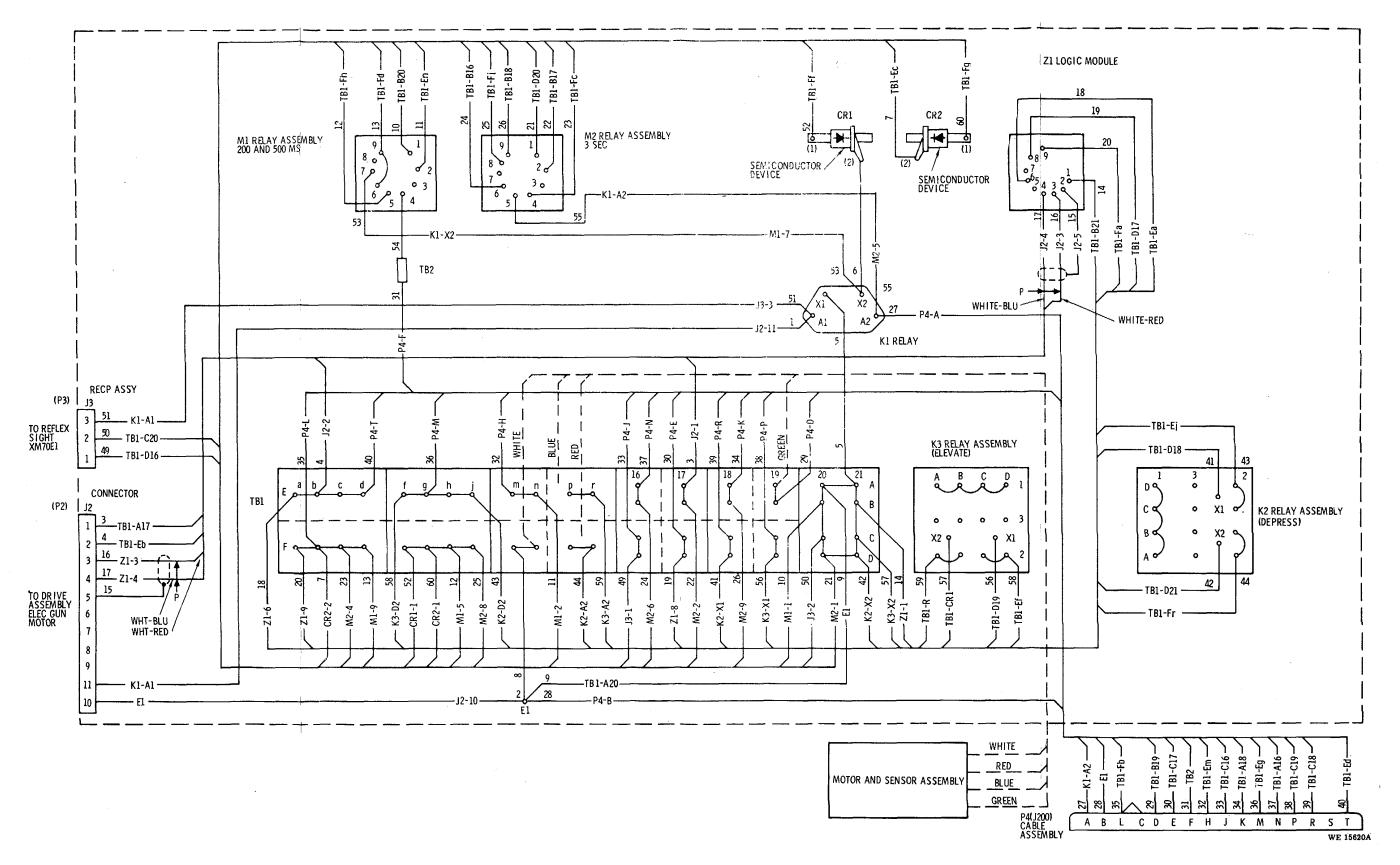


Figure 2-2. Wiring diagram-control box.

Table 2-3. Troubleshooting Electrical System

	1	r			Proc	edure
Malfunction	Probable cause	Action required for cheek	Check between	Normal meter reading (supply voltage 28 VDC)	Meter reading normal	Meter reading abnormal
						-
Note. During electronic operation.	ical troubleshooting	do not operate gun drive	motor except when s	pecifically required. Disc	onnect gun drive cable	plug P1 or P2 to prevent
operation.	1	1	1	1	1	
Gun does not rotate. Fault traced to control box (ref TM 9-1005-298- 12).	Faulty gun contactor K1 or relay M2. tern and hold trig- ger at first detent	Disconnect plug P2, at control box cover, energize sys-	J2-10 and J2-11 Contactor K1-A1 and GRD.	28 VDC 28 VDC	No fault Check wiring from K1-A1 to J2-11.	Continue checks. Continue checks.
,		<i>Note</i> . Burst control relay M2 will open after three seconds. Release trigger	Contactor K1-X2 and GRD.	28 VDC	Replace contactor K1.	Continue checks.
		and depress again to reenergize the firing circuit.	Relay M2-8 and GRD.	28 VDC	Replace diode CR1	Continue checks
			Relay M2-9 and GRD.	28 VDC	Continue checks.	Check box wiring to P4-K and through aircraft wiring to trigger switch.
			Relay M2-4 and GRD.	28 VDC	Replace relay M2 (contacts not closing).	Check box wiring to P4-L and through aircraft wiring to ARMED switch.
Gun rotates at slow rate but does not switch to fast rate. Fault traced to control box.	Faulty relay M2, or trigger switch.	Same as preceding except trigger depressed to second detent	J2-1 and J2-10	28 VDC	No fault	Check box wiring to relay M2, pin 2.
			Relay M26 and GRD.	28 VDC	Replace relay M2 (contacts not closing).	Check box wiring to P4-N and through aircraft wiring to trigger switch.
Gun rotates but ejects live ammo. Fault traced to control box.	Faulty diode CR2	Same as preceding except master switch at FIRE TO CLEAR.	J2-2 and J2-10.	28 VDC	No fault	Check box wiring from J2-2 to CR2.
			Each side of CR2 and GRD. 2-7	28 VDC	No fault	Replace CR2.

				Normal motor reading	Procedure			
Malfunction	Probable cause	Action required for check	Check between	Normal meter reading (supply voltage 28 VDC)	Meter reading normal	Meter reading abnormal		
GUN NOT CLEARED light remains on after firing to clear. (Seven or more	Faulty module Z1	Energize system, depress RESET button on control box then release.	Z1–7 and Z1–8.	No continuity	Check circuit from Z1-8 to P4-E and aircraft cir- cuit for short to ground.	Replace Module Z1.		
rounds ejected; sensor not at fault).			J2–3 and Z1–3.	Continuity	Continue checks	Repair box wiring.		
fault).			J2-4 and Z1-4.	Continuity	Continue checks	Repair box wiring.		
			Z1-6 and GRD.	28 VDC	Continue checks.	Refer to checks for faulty diode CR2.		
			Z1–9 and GRD.	28 VDC	Replace Module Z1.	Check box circuit to P4-L and air- craft to ARMED switch.		
Gun clearing op- eration faulty—too few or too many rounds cleared.	Faulty M1 clearing time relay.	Disconnect plug P2, and energize system in FIRE TO CLEAR with trigger at first detent.	J2–11 and J2–10.	28 VDC with trigger depressed Zero VDC almost imme- diately (200ma) after trigger release.	No fault.	Replace clearing time relay.		
Gun does not elevate (motor not at fault).	Faulty relay K3	Hold pilot's switch at ELEV.	K3-X1 and GRD.	28 VDC	Continue checks.	Check box circuit to P4-P, aircraft wiring, and ELEV. switch.		
			K3-A2 and GRD.	28 VDC	No fault.	Replace relay K3.		
Gun does not depress (motor not at fault).	Faulty relay K2	Hold pilot's switch at DEP.	K2-X1 and GRD.	28 VDC	Continue checks.	Check box circuit to P4-R, aircraft wiring and DEP switch.		
			K2-A2 and GRD.	28 VDC	No fault.	Replace relay K2.		
AMMO LOW indi- cation erratic.	Faulty sensor switch	Disconnect plug P4 at aircraft recep- tacle J200. Depress sensor switch.	P4–D and GRD.	No continuity	Check aircraft cir- cuit by applying ground to J200-D AMMO LOW lamp should light.	Replace sensor switch.		

Table A.	Troubleshooting Electrical SystemContinued
----------	--

					Procedure		
Malfunction	Probable cause	Action required for cheek	Check between	Normal meter reading (supply voltage 28 VDC)	Meter reading normal	Meter reading abnormal	
AMMO LOW indi- cation erratic- continued.		Release sensor switch.	P4-D and GRD.	Continuity	No fault.	Replace sensor switch.	
Reflex sight light operation faulty.	Faulty lamp, wiring or rheostat	Disconnect plug P3 and energize system.	J3-1 and J3-2.	28 VDC	No fault.	Check box circuit from J3-1 to P4-J and aircraft circuit to panel.	
		Rotate rheostat knob CW then CCW. Switch lamp fila- ments and repeat.	P3-1 and P3-2.	Decrease in resistance at full CW. Increased resistance at full CCW.	No fault.	Repeat test using lamp known to be good. Check rheostat Continuity at contacts inside lamp housing.	

Section III. PREEMBARKATION INSPECTION OF MATERIAL IN UNITS ALERTED FOR OVERSEAS MOVEMENT

2-4. General

a. Pre-Inspection Points

Warning: Clear weapon before starting inspection (refer to TM 9-1005-298-12). Point weapon in a safe direction and determine if live rounds are present. Check bore and chamber for obstructions; for example, a bullet in the bore or a ruptured cartridge case in the chamber.

The subsystem must be thoroughly cleaned of all grease, dirt, or other foreign matter that might interfere with is proper function, or obscure the true condition of parts.

b. Inspection Points.

(1) Screw heads must be in serviceable condition, and threads must not be stripped. Internal threads must not be stripped.

(2) Cable assemblies must not have loose or damaged connections, cut or worn insulation, broken wires, kinks, or sharp bends.

(3) Material must be free of burrs, particularly those on functional surfaces.

(4) Parts must not be cracked, bent, distorted or damaged, and must be free of detrimental wear.

(5) Rivets must be tight.

(6) Painted surfaces must be free of bare spots.

(7) Rollers and slides must function smoothly.

(8) Welded or brazed joints must not show signs of separation or failure.

(9) Ammunition boxes and chuting must not have dents or bends that would interfere with the passing of ammunition.

(10) Check for deformed, weak, or broken springs.

(11) All locking devices shall be positive in action, and must not become disengaged due to normal operation and firing.

(12) Instruction, warning, and name plates must be present and secure.

(13) Quick release pins and similar devices must function properly and must not be subject to loss during use or transportation.

Section IV. GENERAL MAINTENANCE

2-6. General

Information and instructions contained herein are provided for personnel performing direct and general support maintenance on the material. Supplemental instructions for organizational personnel are contained in TM 9-1005298-12. In subsequent chapters of this (14) Inspect links for bent or broken tabs and bent segments.

(15) Inspect electrical components for improper functioning, physical damage, and missing parts.

(16) Inspect for burned out lamps and damaged lamp covers and switches.

(17) Inspect electrical connections for damage or corrosion.

(18) Inspect optical parts for cracks, scratches, moisture on inside surface of optical cell in projector assembly, or other damage.

(19) Flexible mount must be capable of movement through full range of elevation/depression without binding.

2-5. Specific

a. 7.62 Millimeter Machine Gun M134

(1) *Bolt assembly.* Track ways must not be nicked, burred, cracked, or galled. Roller must not be worn or damaged.

(2) *Firing bolt head subassembly*. Firing pin hole must not be worn or elongated. Extractor lip must not be broken, cracked, or burred.

(3) *Helical compression spring*. Spring must be free of damage with minimum length of 1.67 inches.

(4) *Firing pin.* Striker and tank must not be worn, broken, or cracked.

(5) *Gun barrel*. Inspect for pitting, scoring excessive wear, and stripping of lands.

(6) *Timing pin.* Engage timing pin. Upon release, pin must retract clear of gun rotor.

b. Weapon Functioning. Manually cycle at least six rounds of M172 dummy ammunition through gun. Moving parts must operate smoothly. Rounds must chamber, extract, and eject without binding or catching and dummy ammunition must not be dented by the operation.

c. Ammunition Feed System. Inspect chute and crossover assembly for bent or broken areas and damaged rollers. Check for damage or defects which restrict flow of ammunition.

technical manual, the main assemblies of the armament subsystem are dis-assembled, inspected, cleaned, replaced or re paired, and assembled. Refer to TM 9-1005-298-12 for removal and installation of the components of the subsystem. The illustrations in this

manual are numbered in the sequence of disassembly.

When assembling, the reverse order of disassembly will be followed unless otherwise instructed. Information for direct and general support maintenance units engaged in repair for return to user is covered herein. Subsequent reference to components being worn and requiring replacement is intended to mean that only those items or mechanisms worn to a degree that affects functioning will be replaced.

2-7. General Repair Instructions- Mechanical

a. Disassembly and Assembly Procedures.

(1) In disassembling a unit, remove the major subassemblies first. The subassemblies may then be disassembled, as necessary, into individual parts.

(2) During assembly, subassemblies should be assembled first and then installed to form a complete unit.

(3) Complete disassembly of a unit is not always necessary in order to make a required repair or replacement. Good judgment should be exercised to keep disassembly and assembly , operations to a minimum.

b. Use of Tools.

(1) Care must be exercised to use tools that are suitable for the task to be performed in order to avoid mutilation of parts and/or damage to tools.

(2) Keep tools clean and work with clean parts. The rules of good housekeeping must be observed.

c. Replacement of Parts.

(1) When pressing bushings out of or into a housing, be sure that the housing is adequately supported in the bearing area to prevent strain and distortion. An arbor press should be used if available. Alternate methods, such as the use of a hardwood block or dowel and a hammer are permissible if necessary. If a bushing has a drilled hole or groove for lubrication, be sure it is clean and properly aligned at installation. Be sure bushings are started properly and not cocked in the bore prior to applying pressure for installation.

(2) When removing or installing ball or roller bearings, the pressing force should be applied only to the race that is restrained by a mating part. Applying force to the unrestrained race will damage the bearing.

(3) During assembly of components, replace all small parts such as spring pins, screws, bolts, and nuts that show signs of wear or damage and might fail prior to the next scheduled maintenance.

Note. Bending or failure of the pins used in assembling rotating components of the gun or delinking feeder can cause major damage due to loss of timing and synchronization.

(4) If a required new part is not available, reconditioning of the old part is required. Such parts should be examined carefully, after reconditioning, to determine that they will function properly.

d. Welding. For welding instructions and welding materials, refer to TM 9-237.

Caution: Welding is not to be accomplished without knowledge of physical characteristics of the metal to be welded.

(1) The link and cartridge ejection chutes are made of corrosion resistant steel. Repairs can be accomplished by spot welding, fusion welding, or brazing.

(2) Most components of the mount assembly are made of sheet or cast aluminum. Welded repairs should not be attempted. Distortion and loss of strength will be caused by welding temperatures.

e. Repair of Damaged Threads. Damage threads should be repaired by use of a thread restorer, by chasing on a lathe, or by use of a tap of proper dimension; damaged or defective inserts will be replaced with oversize inserts.

f. Repair of Damaged Machined and Polished Surfaces. Smooth rough spots, scores, burrs, galling, and gouges from damaged machine and polished surfaces so that the part will efficiently perform its normal function. The finish of the repaired part is to approximate that of the original finish. In performing any of these operations, critical dimensions must not be altered.

g. Removal of Rust or Corrosion. Remove corrosion from all parts of the material. To remove rust or corrosion, the use of crocus cloth, vapor blast equipment, or wipe-on type phosphoric acid metal conditioner is recommended.

h. Finish of Metals.

(1) Painted surfaces of the armament, if chipped or cracked, may be repainted. See TM 9-213.

(2) A class A or class B phosphate finish will be used on ferrous metals unless otherwise specified.

(3) Exposed electrical components will be coated lightly with oil varnish MIL-V-173B.

(4) A type I or II anodic coating will be applied to all aluminum or aluminum alloy parts unless otherwise specified.

(5) It will not be necessary to refinish parts that already have a good quality finish and that refinishing will not definitely improve. Replace damaged decals in accordance with TM 9-213.

(6) All parts will be free from rust, fungus and corrosion.

2-8. General Repair Instructions - Electrical

a. Use wire of same gage and length as that removed. Refer to TM 9-1005-209-35P and to electrical disassembly figures in this manual for wire specifications and terminal or contact part numbers.

b. All soldering ha be in accordance with MIL-STD-454 REQ 5. Use only non-corrosive flux.

c. Use the tools provided in supplemental tool set MOS45J for extraction and insertion of snap-!n contacts in connectors and in the terminal block of the control box.

Caution: Do not attempt to pull out the snap-in contacts without using the proper extraction tool. Damage to connector or terminal block may result.

d. When replacing cable connector wires, use the following typical procedure:

(1) Slip extraction tool into socket cavity from face of connector to compress contact retaining clip. Push wire and contact out of connector. See figure 2-3.

(2) Install new contact on new wire and crimp. (Solder No. 8 contacts).

(3) Use insertion tool to push wire and contact into the connector socket cavity from rear of connector. Hold wire in place while removing insertion tool then check to be sure that contact is locked in the connector by pulling lightly on wire.

e. When replacing components having connections to the terminal block in the control box, remove and insert wire contacts as follows:

(1) Slip extraction tool over the affected wire then push tool into the terminal block socket cavity until contact retaining clip is compressed. Pull tool and wire with contact out of terminal block.

(2) To insert, slip insertion tool over wire and contact retaining clip then push tool and contact into socket cavity. Hold wire in position while withdrawing tool.

2-9. Cleaning

Caution: Do not use solvents or degreasing compounds on electrical components, rubber parts, optical components, or sealed bearings.

a. Cleaning Mount Assembly Components. Components of the mount assembly normally will not require extensive degreasing. The use of dip tank, vapor, or steam cleaning will not be required. These parts can be cleaned using mineral spirits paint thinner (TPM) or dry cleaning solvent (SD) and cloths, soft bristle brushes, or an air pressure spray nozzle.

b. Cleaning Sight Assembly Components. Wipe metal components of the sight with a cloth moistened with solvent (SD) or mineral spirits (TPM), then dry with a clean dry cloth. Clean optical components in accordance with TM 9-208-1 and TM 9-247.

c. Cleaning Parts Requiring Degreasing. Parts of the gun and delinking feeder may be cleaned using one of the three procedures following when extensive degreasing is required.

Caution: Be sure to remove all electrical components and all sealed bearings before using any of the following procedures

(1) *Dip-tank method.* Disassemble as required, place parts in a perforated metal basket, and submerge and agitate in a tank containing dry cleaning solvent of mineral spirits paint thinner. Repeat, using a second tank with clean solvent or thinner. Extent of treatment in each tank will depend on ease with which the preservatives are dissolved.

(2) *Vapor-degreaser method.* Tank containing a heated solution of trichlorethylene or perchlorethylene (type II) are used mostly for degreasing items that are very greasy or oily and are not rapidly cleaned by the diptank method.

Warning: Personnel operating vapor degreaser are cautioned not to breathe the vapor fumes.

Place parts in a perforated metal basket and submerge just below the vapors in the tank and keep there until all the grease or oil melts and runs off the parts in the basket.

(3) *Steam method*. Place parts in a perforated metal basket and steam treat until clean.

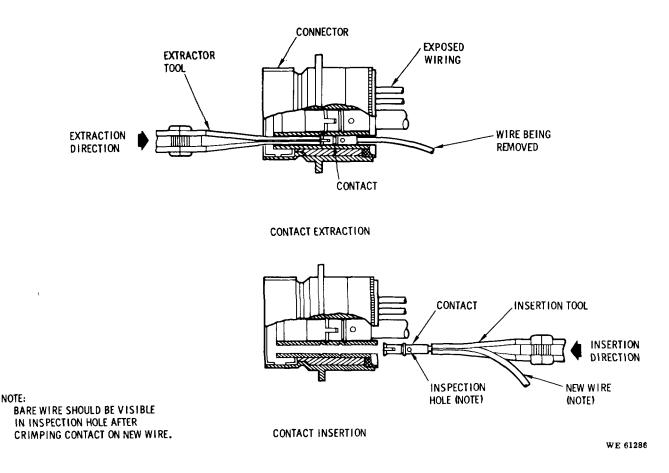


Figure 2-3. Connector contact extraction and insertion.

This method is less efficient than the vapor-degreaser method and may require additional cleaning of parts to remove all traces of grease or oil, particularly from recesses.

(*a*) If some time is to elapse before the start of repair operations, apply light grade of preservative oil to all polished metal surfaces to prevent rusting.

(*b*) Remove all rust spots from highly finished surfaces with a light application of crocus cloth. Use a grade 2/0 abrasive cloth on ordinary machine finished surfaces.

d. Cleaning Gun and Delinking Feeder After Repair.

(1) After repair operations and prior to assembly, remove shop dirt and other foreign matter from all metal surfaces. This can be done by the diptank method, the vapor-degreaser method, or by cleaning with cloths soaked in dry-cleaning solvent or mineral spirits paint thinner.

(2) In the dip-tank method, agitation for approximately one minute in each tank is sufficient; in the vapor-degreaser method, treatment for about two to three minutes is sufficient.

e. Cleaning Gun and Delinking Feeder After Shop Inspection. After in-process shop inspections, dip parts in a tank containing fingerprint remover oil (type A), remove (use rubber gloves), and dry thoroughly by wiping with clean, lint-free, dry cloths or allow parts to drip dry. Apply preservatives as soon as possible after cleaning.

f. Cleaning Electrical Parts. Clean all electrical parts in accordance with TM 9-247.

g. Cleaning Rubber Parts Other Than Electrical. Clean rubber parts with soap and warm water. Apply coating of powdered technical talcum (Fed. Spec. ZZ-T-416) to preserve the rubber.

h. Lubrication and Preservation.

(1) Lubrication. Refer to LO 9-1005-298-12 for lubricating instructions for the subsystem. Lubrication that is performed during assembly is noted in the applicable assembly procedures. (2) *Preservation.* After cleaning and drying, immediately coat unpainted metal surfaces of gun and

delinking feeder with lubricating oil (LSA-T) in accordance with LO 9-1005-298-12.

Section V. REMOVAL AND INSTALLATION OF MAJOR COMPONENTS

2-10. Removal/Installation

Refer to TM 9-1005-298-12 for applicable procedures.

REPAIR OF GUN ASSEMBLY

Section I. GUN ASSEMBLY

3-1. Disassembly/Assembly

The gun assembly is disassembled and assembled at organizational level. Refer to TM 9-1005-298-12.

Individual major components requiring further disassembly and assembly are covered in subsequent sections of this chapter.

Section II. 7.62 MILLIMETER MACHINE GUN M134

3-2. Disassembly/Assembly

The gun is disassembled in the order of item numbers shown on figures 3-1, 3-2, and 3-3. Assembly is performed in reverse order. Read the general repair instructions for mechanical components (para 2-7) prior to start of work.

3-3. Cleaning, Inspection, and Repair

a. Cleaning. Refer to paragraph 2-9.

b. Inspection. Refer to general repair instruction table in TM 9-10(-298-12.

c. Repair. Refer to general repair instructions for mechanical components (para 2-7).

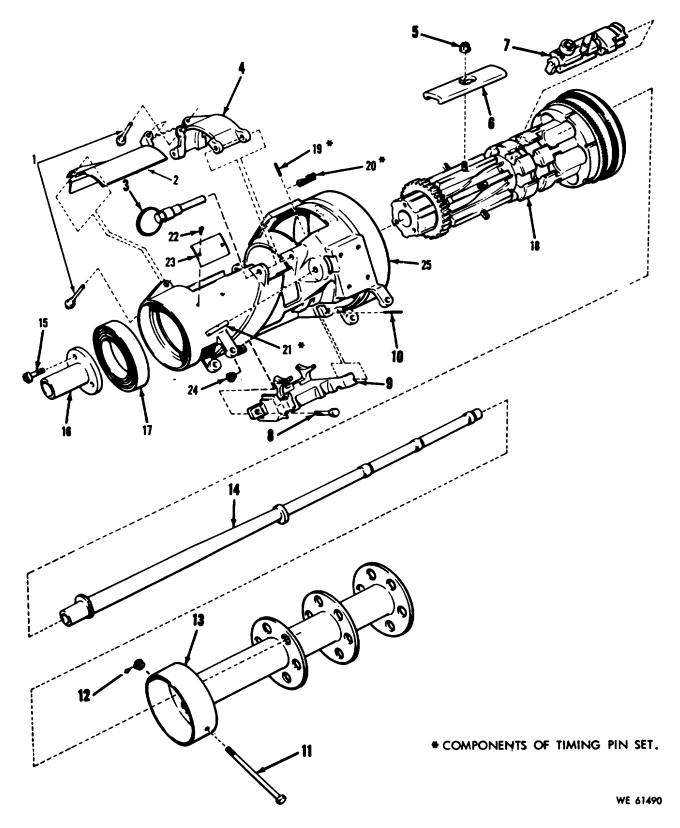


Figure 3-1. 7.62 Millimeter machine gun M134 exploded view.

1-Pin 2-Cover 3-Pin 4-Safing sector 5-Nut 6-Track 7-Bolt assembly 8-Screw 9-Bar 10-Pin 11-Bolt 12-Splined nut 13-Barrel clamp 14-Barrel 15-Screw 16-Support 17-Bearing 18-Rotor assembly

Figure 3-1—Continued.

3-3

19-Pin 20-Spring 21-Pin 22-Drive screw 23-Identification plate 24-Insert 25-Gun housing

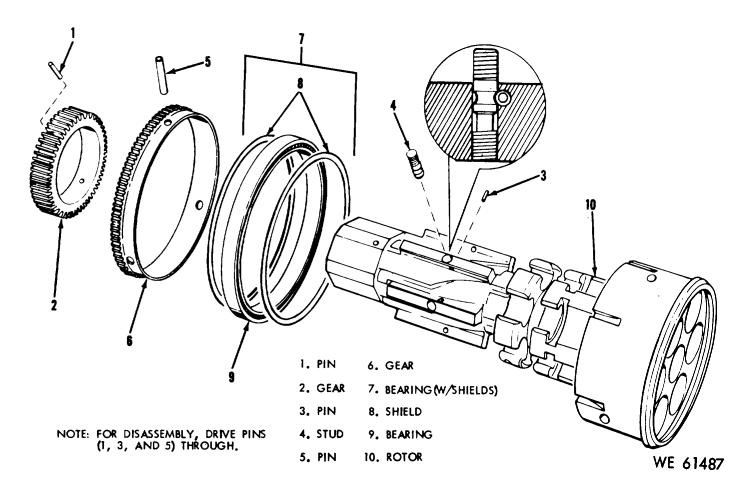


Figure 3-2. Rotor assembly-exploded view.

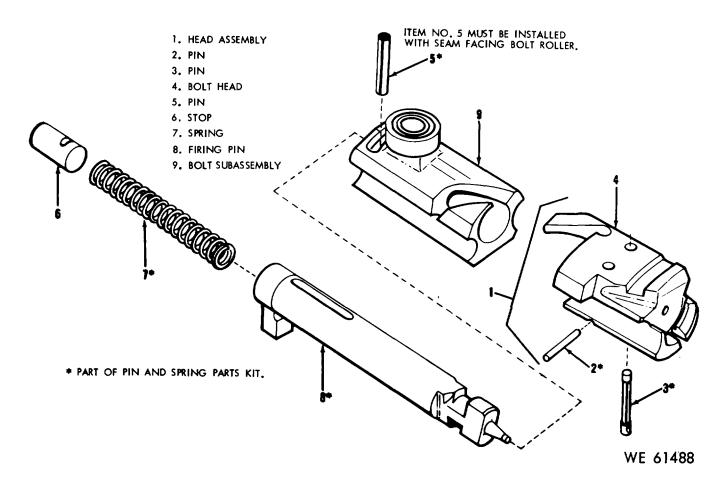


Figure 3-3. Bolt assembly-exploded view.

Section III. DELINKING FEEDER MAU-56/A

3-4. Description

The delinking feeder provides a flow of delinked ammunition to the gun and is powered, through gearing, by the gun. Additional descriptive information will be found in TM 9-1005-29812.

3-5. Disassembly/Assembly

The delinking feeder is disassembled in the order of item numbers shown on figure 3-4. Assembly is performed in reverse order. Read the general repair instruction for mechanical components (para 2-7) prior to start of work.

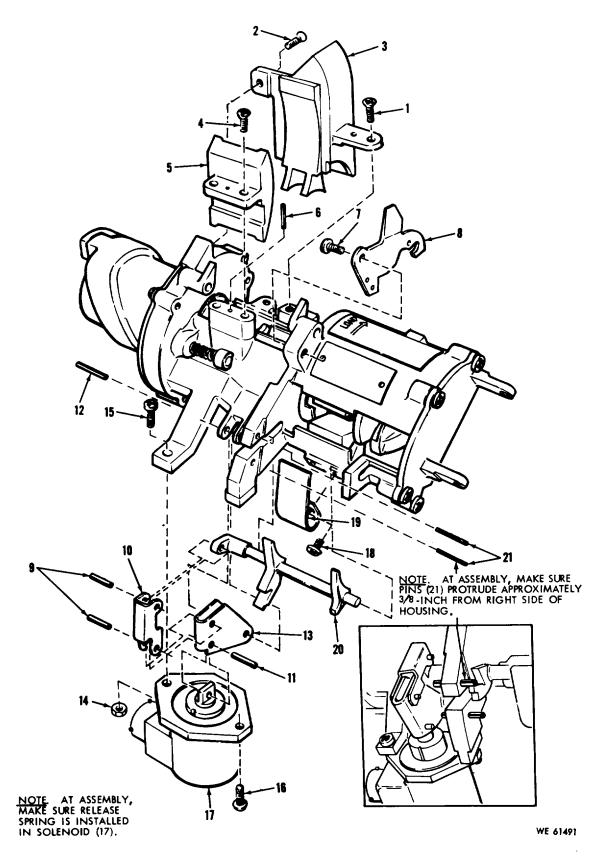


Figure 3-4. Delinking feeder MAU-56/A-exploded view (sheet 1 of 3).

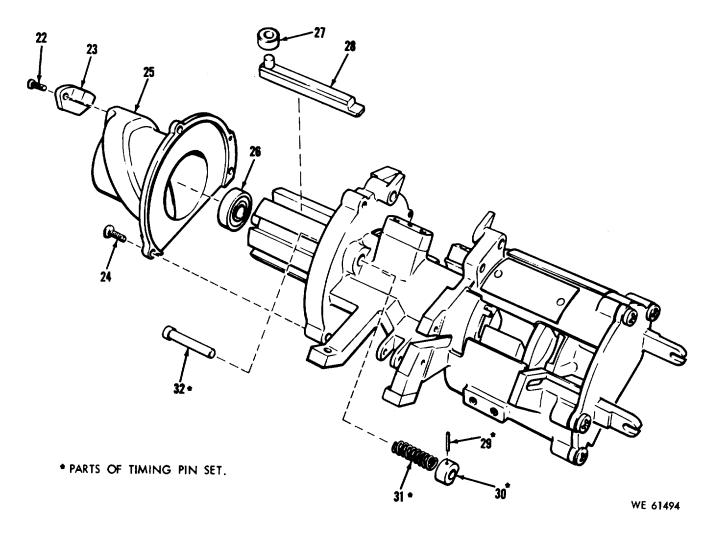


Figure 3-4. Delinking feeder MA U-56/A—exploded view (sheet 2 of 3).

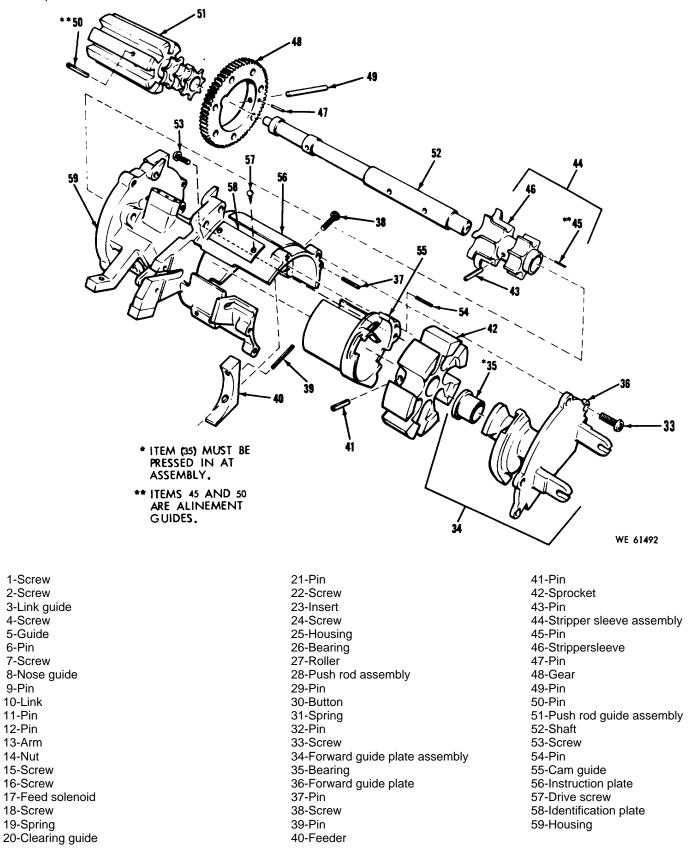


Figure 3-4. Delinking feeder MAU-56/A—exploded view (sheet 3 of 3).

3-6. Cleaning, Inspection, and Repair

a. Cleaning. Refer to paragraph 2-9.

b. Inspection. Check for worn bearings; check all parts for galling chipping, and metal transfer, check for

Section IV. GUN ELECTRIC DRIVE ASSEMBLY

3-7. Description

The gun electric drive assembly consists of two major components, the gun drive control assembly and the electric drive assembly. The gun drive control assembly, which controls gun speed, is replaceable at organizational level. Refer to TM 91005-298-12. The electric drive assembly consisting of a 28-volt DC motor and a gear housing, requires further disassembly and assembly as shown in figures 3-5 and 3-6.

Caution: The motor and the gear housing are matched assemblies having identical serial numbers. Do not interchange nonmatching units. Disassembly of the motor, other than as shown, is not recommended. The motor housing, end bells, and armature are matched and electrically aligned by the manufacturer.

3-8. Disassembly

Refer to TM 9-1005-298-12 for information regarding removal of the gun drive control assembly from the electric drive assembly. Disassemble the electric drive assembly as shown in figure 3-5 and in the sequence of item numbers in figure 3-6. Observe the following special requirements.

a. When disassembling the gearing from the gear housing (3, fig. 3-6), it will be necessary to drill a hole (approx 1/8-inch diameter) in retaining cap (4). Pry the cap out with a small punch or similar tool. Discard the damaged cap.

bent or sheared spring pins and aligning pins.

c. Repair. Refer to general repair instructions for mechanical components (para 2-7).

b. Make note of the number of shims (17) used under the ball bearing (16) on end of shaft (9).

Caution: Do not remove needle bearings from gear housing unless in-place inspection reveals a fault. Damage can be caused by removal. Refer to table 3-1.

3-9. Cleaning, Inspection, and Repair

a. Cleaning. Clean needle bearings, gears, shafts, and gear housing with solvent (SD or TPM).

Caution: Do not immerse sealed ball bearings or the motor assembly in solvent. Wipe ball bearings with a clean cloth and use an air pressure nozzle to blow carbon brush dust out of motor.

b. Inspection. Refer to table 3-1.

c. Repair. Repairs are limited to replacement of parts provisioned; refer to TM 9-1005-298-35P. Observe the following repair requirements.

(1) Replace all needle bearings that are removed. These bearings are sized to the mating shaft by being pressed into the gear housing and are usually damaged by removal.

(2) Replace brush springs (fig. 3-5) whenever brushes are replaced. Springs may have lost temper and strength though no visible fault exists.

(3) Observe Caution in paragraph 3-7 against interchange or replacement of matched components.

Component	Figure and item	Inspection for
All gears	3-6, 7, 10, 12, 14	Chipped or broken teeth and evidence of galling, metal
Needle bearings	3-6 21, 21, 23	transfer, or excessive wear. (Wear through the black phosphate coating is normal.) Radial play on shaft prior to shaft removal. Evidence of
needie beanings	3-0 21, 21, 23	bearing roughness when mating shaft is turned. (Bear- ings are sized for correct shaft fit when pressed into .the gear housing and may be oversize or damaged when removed.)
Ball bearings	3-6, 18, 20	Axial (end) play between inner and outer races. Evidence of roughness felt when the bearing is turned.
Shafts	3-6 6, 9	Breaks, galling, or evidence of ball bearing rotation on shaft.
Armature	3-5	(Visually through brush access openings.) Pitting, scoring, or excessive wear of commutators. Evidence of over-
		3-9

Table 3-1. Electric Drive Assembly Inspection Points

Component	Figure and item	Inspection for
		heating such as thrown solder at commutator segment
		connections or blackened commutators. (Normal color is
		copper/brown.)
Motor housing	3-5	External damage. All visible wiring for breaks, worn insulation, or loose terminals
Brush holders	3-5	(Visually through brush access openings.) Breaks, bends, stripped screw threads, or looseness in endbells (epoxy retaining material chipped out.)
Brushes	3-5	Breaks, chipping, loose wire or terminal, and excessive wear (0.25-inch min. length from shoulder to face).
Brush springs	3-5	Breaks and deformation.

3-10. Assembly

Assemble the electric drive assembly as shown in figure 3-5 and in essentially the reverse of item number sequence shown on figure 3-6. Observe the following special requirements:

a. When installing new brush springs, note that the springs are shaped at one end to fit tightly and be retained on the guide pins in the brush covers.

b. Press in new needle bearings flush with respective inside shoulders of the gear housing. Coat the rollers of these bearings with grease (GIA). Assemble shafts, gears and ball bearings as shown in sectional view on figure 3-5. Apply grease (GIA) on gear teeth but do not "pack" the gear housing.

c. When installing the gear housing assembly on the motor forward end bell, use the same number of shims under the output shaft bearing as were removed during disassembly. Check the shaft for 0.003 to 0.008-inch end play. Remove the gear housing, add or remove shims, and reinstall gear housing if required.

3-11. Gun Electric Drive Assembly Testing

The electric gun drive assembly may be tested for proper operation by installing it on the subsystem as follows:

Note Extensive running for wear-in of replacement parts is not required.

a. Install gun drive control assembly on the electric drive assembly and install the assembly on a mounted gun. Refer to TM 9-1005-298-12 for procedure.

Warning: Be sure there is no ammunition present in any subsystem component before operating the gun drive motor.

b. Follow the procedure for operational check with power on (TM 9-100-212) except as follows:

(1) Operate the gun at low speed (first trigger detent) for three 3-second bursts. Observe gun barrel rotation.

Caution: High speed operation of the gun should be limited to momentary bursts. Gun may be damaged by over-speeding when operated without ammunition.

(2) Check operation at high speed by pressing the trigger to the first detent, then momentarily to the second detent. Observe that gun barrel rotational speed increases.

c. Follow the steps of operational check procedure (TM 9-1005-29812) required to safe the subsystem and remove all electrical power.

d. Remove one brush cover and one brush from each armature commutator position. Turn the armature by rotating the gun barrels and inspect the armature for evidence of excessive arcing or overheating. Reinstall brushes and brush covers.

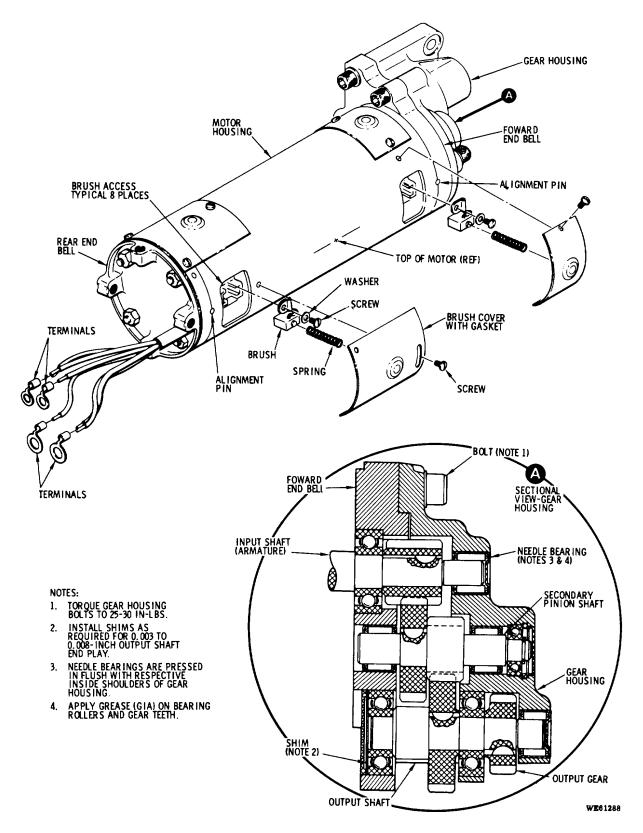


Figure 3-5. Electric drive assembly—partially exploded view.

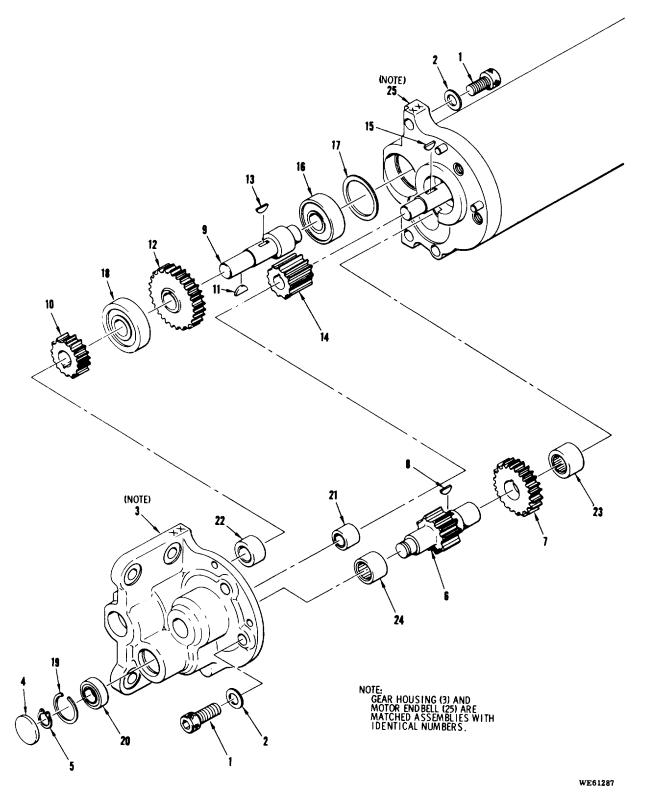


Figure 3-6. Electric drive assembly gear housing—exploded view.

1-Bolt 2-Washer 3-Gear housing 4-Retaining cap 5-Snap ring (retainer) 6-Pinion 7-Gear 8-Key 9-Shaft 10-Pinion gear 11-Key 12-Gear 13-Key 14-Pinion gear 15-Key 16-Ball bearing 17-Shim 18-Ball bearing 19-Snap ring (retainer) 20-Ball bearing 21-Needle bearing 22-Needle bearing 23-Needle bearing 24-Needle bearing 25-Forward end bell

Figure 3-6—Continued.

Section V. SENSING UNIT AND CABLE ASSEMBLY

3-12. Disassembly/Assembly

See figure 3-7.

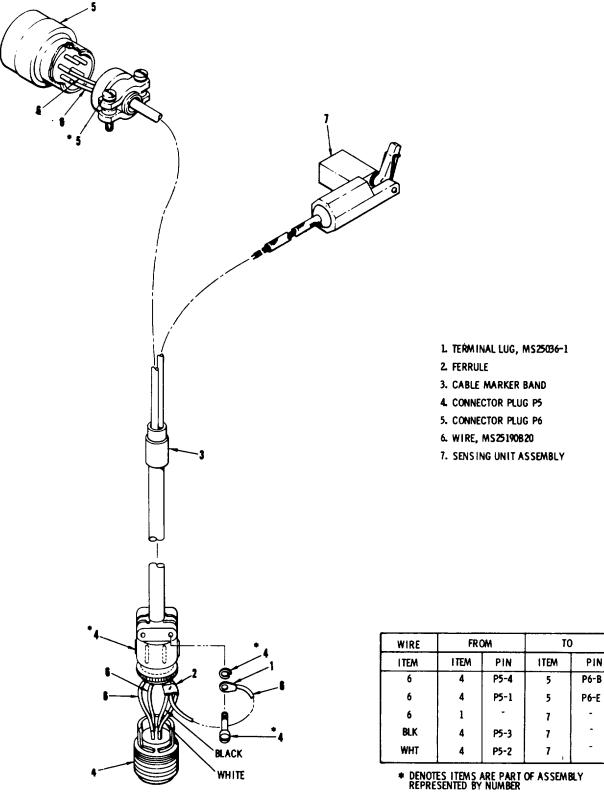
3-13. Cleaning, Inspection, and Repair

a. Cleaning. Refer to TM 9-247 for cleaning instructions.

b. Inspection. Inspect for frayed insulation, broken wires, and loose connections. Inspect connectors for

bent or broken pins and corrosion at terminals. Isolate faulty components by using troubleshooting procedures on table 2-3.

c. Repair. Repair cable in accordance with general procedures in paragraph 2-8. Refer to table on figure 3-7 and to TM 9-1005-298-35P for wire specifications and for connector and contact part numbers.



WE 15602

Figure 3-7. Sensing unit and cable assembly.

CHAPTER 4

REPAIR OF FAIRING ASSEMBLY

4-1. Description

The fairing assembly is constructed on mat-molded fiberglass which is resin impregnated and cured while under heat and pressure. The fairing assembly is important to the operation of the armament subsystem XM27E1 since the upper fairing assembly with its ram air duct provides for accelerated ejection of cartridge links. This prevents the links from striking the aircraft during inflight firing.

4-2. Disassembly/Assembly

See figure 4-1. Limit disassembly to removal of faulty components such as fasteners. receptacles, and decals.

4-3. Cleaning, Inspection, and Repair

a. Cleaning. Refer to paragraph 2-9a and apply procedures used for cleaning mount assembly components.

b. Inspection. Check for damaged fasteners and receptacles and for breaks and tears in the fiberglass structure.

c. Repair.

(1) Replace damaged fasteners and receptacles using standard shop procedures and tools.

(2) Repair breaks and tears in fiberglass structure as follows:

(a) Stop-drill the ends of the break or tear using a 1/8-inch (No. 30) drill.

(b) Lay out and pencil mark an area extending approximately one inch beyond all edges of the damage.

Warning: If a power sander is used, avoid breathing the fine fiberglass dust produced. This dust can cause serious skin and respiratory irritations. Wear protective clothing and use a mask.

(c) Use 280 grit sandpaper to sand the area to be patched. Sanding should be deepest at the center of the damaged area and beveled out at the edges.

(d) Clean the area to be patched with methyq-ethyl-ketone, Specification TT-M-261.

(e) Cut a fiberglass patch of appropriate size to fit into the beveled area to be patched. Fray all edges to fair in smoothly.

(f) Mix resin and catalyst in the proportions recommended on the manufacturer's instruction sheet.

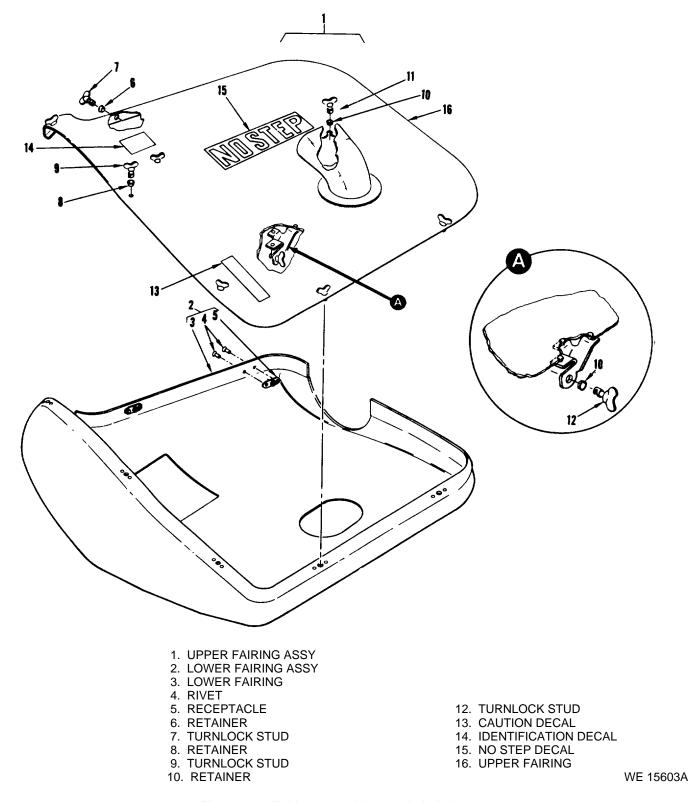
(g) Apply a coat of the resin to the area to be patched, place fiberglass patch in position and apply a heavy second coat of resin to the top of the patch. The top coat of resin should extend sufficiently beyond the edges of the patch to assure a smooth feathered edge.

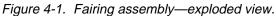
(h) Tape a piece of cellophane over the patch and extending 2 or 3 inches beyond the patch edges.

(i) Gently work out all air bubbles and shape the top coat of resin to fair in smoothly with the contours of the fairing.

(j) Subject the patch to curing heat for the time specified by the manufacturer's instructions.

(k) Remove cellophane, sand patch lightly, then paint and replace any damaged decals (TM 9-213).





CHAPTER 5

REPAIR OF MOUNT ASSEMBLY

Section I. MOUNT ASSEMBLY

5-1. Disassembly/Assembly

Major components of the mount assembly are disassembled and assembled as shown in figure 5-1.

5-2. Door Filler Assembly

Replaceable components of the flexible plastic door filler assembly are shown on figure 5-2. No special

maintenance information is required.

5-3. Control Rod Assembly

The control rod assembly is maintained at organizational level and all replacement parts are provisioned at that level. Refer to TM 9-1005-298-12.

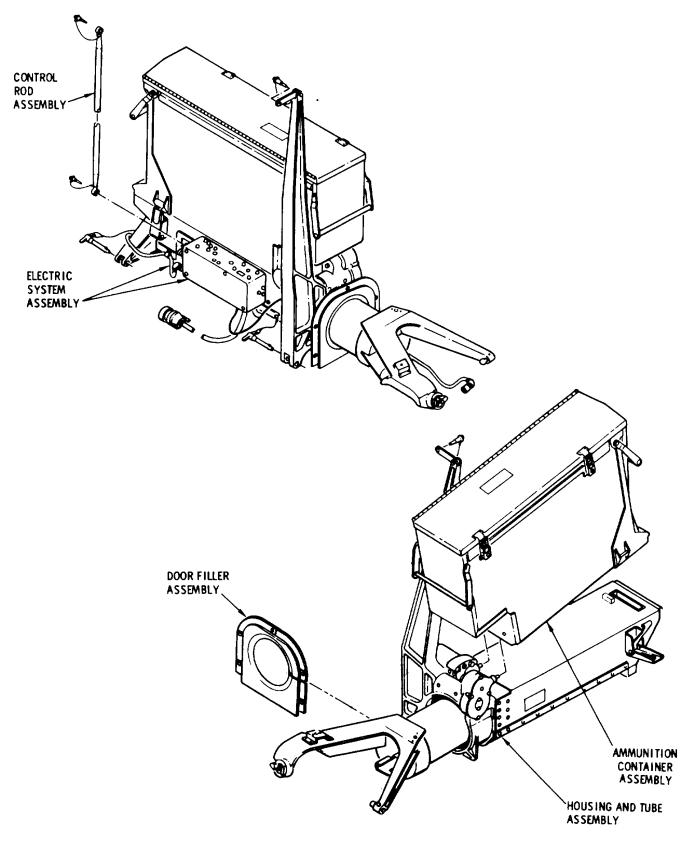


Figure 5-1. Mount assembly—exploded view.

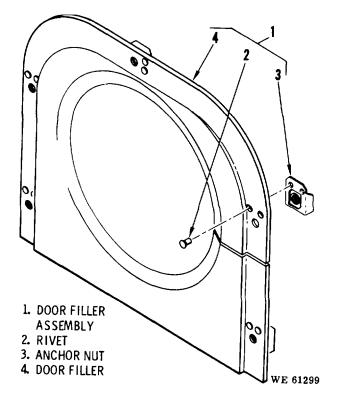


Figure 5-2. Door filler assembly—exploded view.

Section II. AMMUNITION

5-4. Disassembly/Assembly

Disassemble the ammunition container assembly as required for replacement of worn or damaged parts in the order of item numbers on figure 5-3. Reassemble in reverse of item number order.

5-5. Cleaning, Inspection, and Repair

a. Cleaning. Refer to paragraph 2-9a for general cleaning instructions.

b. Inspection.

(1) Check for worn or damaged roller assemblies (31, fig. 5-3).

(2) Check box cover, handles (3), and levelers (23), for breaks and bends.

(3) Check levelers for broken springs (22).

(4) Check box structure for dents and breaks.

c. Repair. Repair by replacing worn or damaged parts and by straightening dents and bends when feasible.

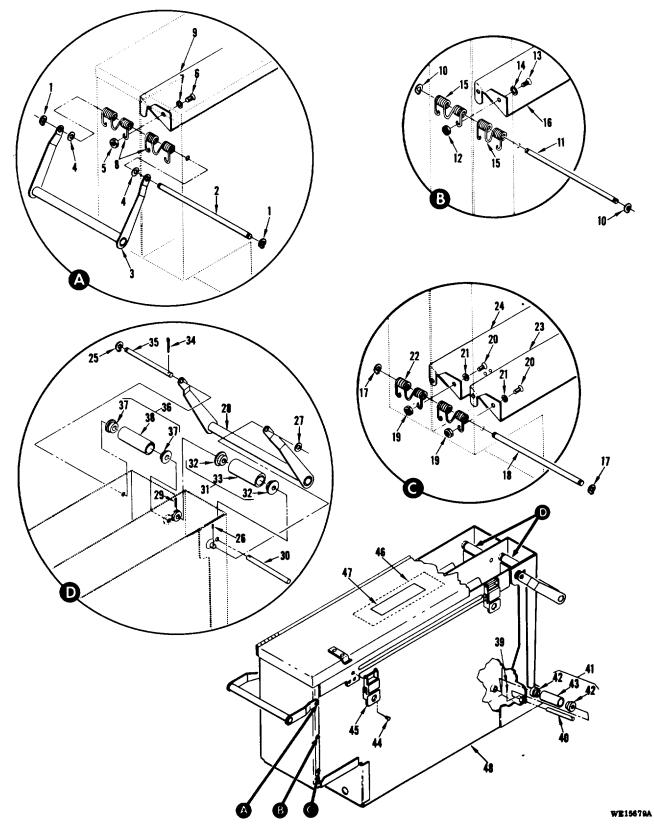


Figure 5-3. Ammunition container assembly—exploded view.

1-Retaining ring 2-Pin 3-Handle 4-Flatwasher 5-Nut 6-Machine screw 7-Flat washer 8-Spring 9-Leveler 10-Retaining ring 11-Pin 12-Nut 13-Machine screw 14-Flat washer 15-Spring 16-Leveler

17-Retaining ring 18-Pin 19-Nut 20-Machine screw 21-Flat washer 22-Spring 23-Leveler 24-Leveler assembly 25-Retaining ring 26-Spring pin 27-Flat washer 28-Handle 29-Cotter pin 30-Pin 31-Roller assembly 32-Sleeve bearing

33-Sleeve spacer 34-Cotter pin 35-Pin 36-Roller assembly 37-Sleeve bearing 38-Sleeve spacer 39-Cotter pin 40-Pin 41-Roller assembly 42-Sleeve bearing 43-Sleeve spacer 44-Rivet 45-Rim latch 46-Loading decal 47-Caution decal 48-Ammunition container

Figure 5-3—Continued.

Section III. ELECTRICAL SYSTEM ASSEMBLY

5-6. Disassembly/Assembly

The electrical system is disassembled and reassembled as shown on figure 5-4.

a. Control Box Subassembly. The control box subassembly is disassembled and assembled as shown in figure 5-5.

b. Special Purpose Branched Cable. The special purpose branched cable is disassembled and assembled as shown in figure 5-6.

5-7. Cleaning, Inspection, and Repair

a. Cleaning. Electrical system components normally will not require cleaning. Solvent or cleaning fluids should not be used due to possibility of damage to insulation or impairment of electrical continuity. Wipe out any foreign matter with a clean cloth or blow out using an air pressure nozzle.

b. Inspection. Inspect electrical system for frayed insulation, broken wires, and loose connections. Inspect connectors for bent or broken pins and corrosion at terminals. Isolate faulty components by using troubleshooting procedures of table 2-3.

c. Repair. No repair of components within the control box is feasible. Replace those found faulty. Repair wiring and cables in accordance with general procedures in paragraph 2-8. Refer to TM 9-1005-29835P and to wire list tables for wire specifications and for connector and contact part numbers.

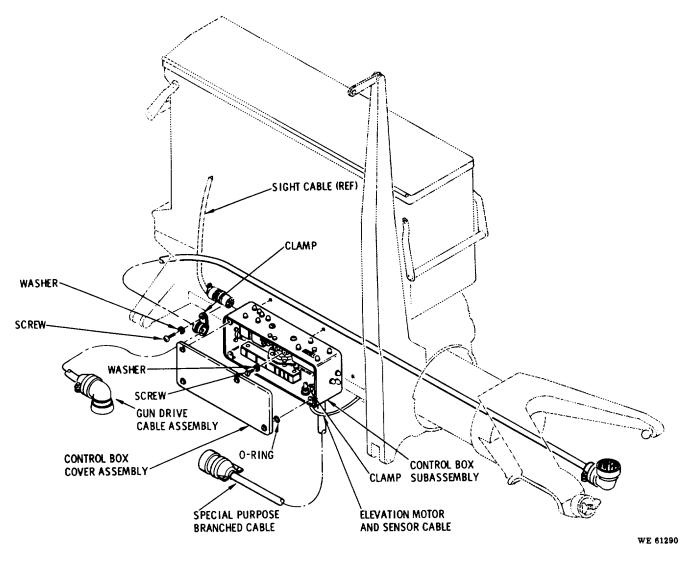


Figure 5-4. Electrical system assembly.

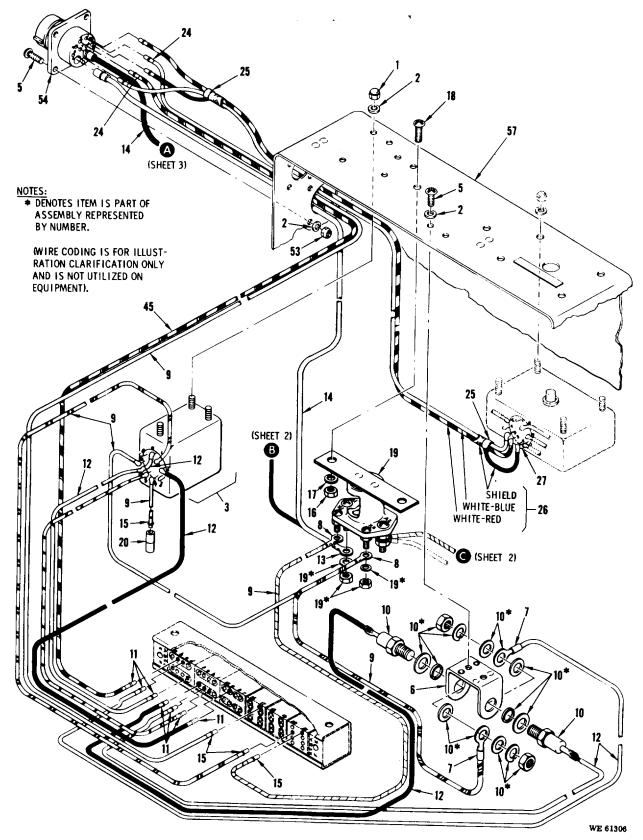


Figure 5-5 (1). Control box subassembly—partial exploded view (sheet 1 of 3).

1-Nut 2-Washer	20-Splice, 11 21-Nut
3-Relay assembly M1	22-Washer
4-Relay assembly M2	23-Relay ass
5-Screw	24 Contact, 7
6-Bracket assembly	25-Sleeving
7-Insulation	26-Wire, AW
8-Terminal lug, MS25036-1	27-Wire, QQ
9-Wire MS25190B20	28-Logic mo
10-Diodes, CR1 and CR2	29-Insulation
11-Contact 11697831	30-Relay ass
12-Wire, MS25190B16	31-Nut
13-Terminal lug, MS25036-15	32-Washer
14-Wire, MS25190B18	33-Screw
15-Contact, 11697830	34-Loop clar
16-Nut	35-Loop clar
17-Washer	36-Screw
18-Screw	37-Bracket
19-Relay K1	38-Grommet
5	

- 20-Splice, 11697579 21-Nut 22-Washer 23-Relay assembly 24 Contact, 11697840 25-Sleeving 26-Wire, AWG22, MIL-C-7078 27-Wire, QQ-W-343 28-Logic module Z1 29-Insulation 30-Relay assembly K2 and K3 31-Nut 32-Washer 33-Screw 34-Loop clamp 35-Loop clamp 36-Screw 37-Bracket 38-Grommet
- 39-Cable assembly 40-Grommet 41-Screw 42-Terminal lug, MS20659-7 43-Terminal lug, MS25036-8 44-Terminal lug, MS25036-3 45-Wire, MS25190B18 46-Screw 47-Terminal box 48-Module (part of TB1) 49-Module (part of TB1) 50-Module (part of TB1) 50-Module (part of TB1) 51-Module (part of TB1) 52-Identification plate 53-Nut 54-Connector receptacle J2 55-Receptacle assembly J3 56-Connector 57-Control box

Figure 5-5-Continued.

Table 5-1.	Wire List for Figure 5-5	
------------	--------------------------	--

Wire		From		То			
ltem	Item	Pin	Terminal	Item	Pin	Terminal	
9	54	J2-1	24	49	TB1-A17	15	
45	54	J2-2	24	51	TB1-Eb	11	
14	54	J2-10	42	41	GRD	42	
14	54	J2-11	13	19	K1-A1	10	
9	10	CR1-2	10	19	K1-X2	10	
12	19	CR2-2	10	51	TB1-Ec	11	
9	19	K1-X1	8	48	TB1-A21	15	
12	41	GRD	44	50	TB1-Fm	11	
12	41	GRD	43	48	TB1-A20	15	
9	56	J3-1	*	49	TB1-D16	15	
9	56	J3-2	*	48	TB1-C20	15	
9	56	J3-3	*	19	K1-A1	45	
9	8	M1-1	*	48	TB1-B20	15	
9	8	M1-1	*	20	TB2	20	
9	8	M1-5	*	51	TB1-Fh	11	
9	3	M1-7	*	19	K1-X2	8	
12	8	M1-6	*	3	M1-9	*	
12	8	M1-9	*	51	TB1-Fd	11	
12	3	M1-2	*	50	TB1-En	11	
9	4	M2-1	*	48	TB1-D20	15	
9	4	M2-2	*	49	TB1-B17	15	
9	4	M2-5	*	19	K1-A2	44	
9	4	M2-6	*	49	TB1-16	15	
9	4	M2-9	*	49	TB1-B18	15	
9	4	M2-4	*	51	TB1-Fc	11	
9	4	M2-8	*	51	TB1-Fj	15	
12	10	CR1-1	*	51	TB1-Ff	11	
12	10	CR2-1	*	51	TB1-Fg	11	
9	28	Z1-1	*	48	TB1-B21	15	
9	28	Z1-9	*	61	TB1-Fa	11	
9	28	Z1-6	*	61	TB1-Ea	11	
9	28	Z1-8	*	49	TB1-D17	11	
9	28	Z1-2	*	54	J2-5	*	
27	28	Z1-2	*	28	Z1-1	*	
26	28	Z1-4	*	54	J2-4	24	
*Solder c	onnection						
-			5.	-9			

Wire	From				То		
Item	Item	Pin	Terminal	Item	Pin	Terminal	
26	28	Z1-3	*	54	J-3	24	
9	54	J2-5	*	25	SHLD	*	
9	30	K2-X1	*	49	TB1-D18	15	
9	30	K2-X1	*	48	TB1-D21	15	
12	30	K2-A2	*	50	TB1-Fr	11	
12	39	K2-D2	*	51	TB1-Ej	15	
9	30	K3-X1	*	49	TB1-D19	15	
9	30	K3-X2	*	48	TB1-C21	15	
9	30	K3-X2	*	50	TB1-Er	11	
12	30	K3-D2	*	51	TB1-Ef	11	
*Soldered	connection						

Table 5-1. Wire List for Figure 5-5

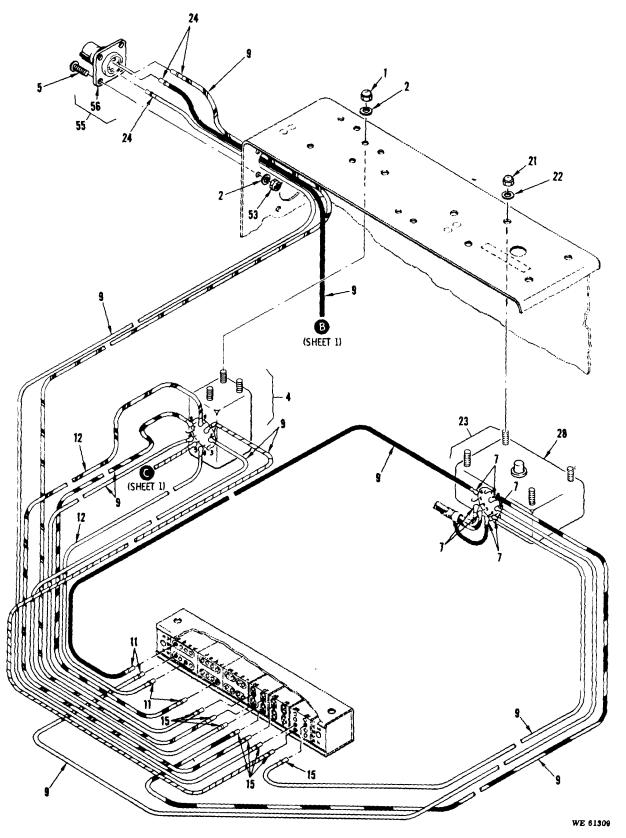


Figure 5-5 (2). Control box subassembly—partial exploded view (sheet 2 of 3).

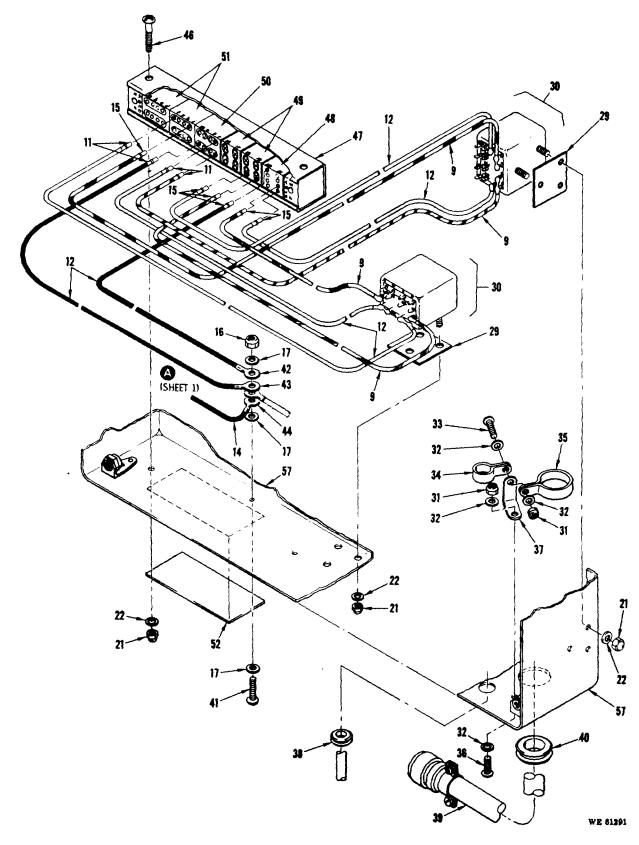


Figure 5-5 (3). Control box subassembly partially exploded view (sheet 3 of 3).

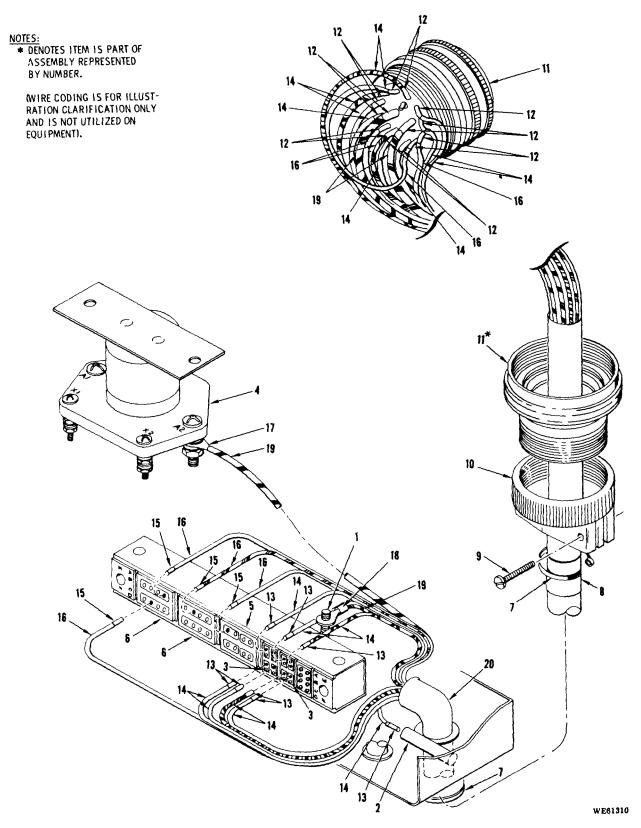


Figure 5-6. Special purpose branched cable assembly.

1-Screw 2-Splice, 11697579 3-Module (part of TB1) 4-Relay, K1 5-Module (part of TB1) 6-Module (part of TB1) 7-Cable marker 8-Cable marker 9-Screw 10-Connector damp 11-Connector plug P4 12-Insulation 13-Contact 11697830 14-Wire, MS25190A20 15-Contact, 11697831 16-Wire, MS25190A16 17-Terminal lug, MS25036-15 18-Terminal lug, MS20669-7 19-Wire, MS25190B8 20-Tubing

Table 5-2.	Wire List	for Figure	5-6
------------	-----------	------------	-----

Wire	From				То		
Item	Item	Pin	Terminal	Item	Pin	Terminal	
14	11	P4-C	*	11	P4-L	13	
14	11	P4-D	*	3	TB1-B19	13	
14	11	P4-E	*	3	TB1-C17	13	
14	11	P4-F	*	2		13	
14	11	P4-J	*	3	TB1-C16	13	
14	11	P4-K	*	3	TB1-A18	13	
14	11	P4-N	*	3	TB1-A16	13	
14	11	P4-P	*	3	TB1-C19	13	
14	11	P4-R	*	3	TB1-C18	13	
16	11	P4-H	*	5	TB1-Em	15	
16	11	P4-L	*	6	TB1-Fb	15	
16	11	P4-M	*	6	TR1-Eg	15	
16	11	P4-T	*	6	TB1-Ed	15	
19	11	P4-A	*	4	K1-A2	17	
20	11	P4-B	*	1	GRD	18	
*Soldered	connection						

Section IV. HOUSING AND TUBE ASSEMBLY

5-8. Disassembly/Assembly

Disassemble the housing and tube assembly in the sequence of item numbers shown in figures 5-7 and 5-8. Further disassembly of the elevation motor and sensor assembly and the ammunition chutes is shown on figures 5-9 and 5-10. Assembly is performed in the reverse order of the disassembly sequence. Observe adjustment and test requirements (para 5-10) during the assembly process and also make use of all applicable general repair instructions (para 2-7).

5-9. Cleaning, Inspection, and Repair

a. Cleaning. Clean in accordance with general cleaning instructions given in paragraph 2-9a.

b. Inspection.

(1) Check housing for dents, breaks, and other damage.

(2) Check bearings, shafts, and gearing for wear and damage.

(3) Check ammunition chutes for breaks, dents, and damage that could restrict flow of ammunition. Check roller and shaft in elbow chute for damage and wear.

(4) Check tube assembly components including shock isolator (11, fig. 5-8), fairing attachment bracket (14), support (16), receptacle (18), bolt ball (2), and camlock (9) for excessive wear or damage.

5-10. Assembly Adjustment and Test Requirements

The following adjustments and tests are to be performed during assembly of the worm gear and clutch and the torque tube and housing

a. Using bearing housing wrench 11697522 (fig. 5-11), screw forward bearing housing assembly (39, fig. 5-7) into mount housing until it bottoms.

b. Adjust screws in bearing housing gage 11697521 (fig. 5-11) for light friction then retract bearing housing gage into torque arm bearing. Using dowel as a lever, slide bearing housing gage up into casting as far as possible.

c. Slide forward bearing housing wrench, 11697522, through the housing and through the bearing housing gage to mate with the forward bearing housing. Turn the wrench counterclockwise to back the bearing housing outward and into contact with the bearing housing gage.

d. Lock the bearing housing in place with set screw (34, fig. 5-7). Seal the set screw with sealant, Military Specification MIL-S-22473, Grade H.

e. Remove forward bearing housing wrench. Retract bearing housing gage 11697521 and remove gages from housing assembly and secure for future use.

f. Slide worm gear (38, fig. 5-7) into housing assembly. Match polarity of worm gear (38) and gear sector on tube assembly (24), "A" end to "A" end and "B" end-to "B" end. If more than one unit is being overhauled, match serial numbers of worm to gear sector.

g. Position aft bearing assembly housing (35) and key (37) on shaft assembly (36). Slide shaft assembly with key into worm gear (38) and engage forward bearing housing assembly (39). Screw aft bearing housing assembly (35) into housing assembly (46). Using bearing housing wrench (11697522, fig. 5-11) tighten aft bearing housing assembly to 125 inch-pounds torque and then back off 15 to 22.5 degrees (1 and 1 1/2 marks on wrench shaft).

h. Lock aft bearing housing assembly (35, fig. 5-7) in place with setscrew (34). Seal the set screw with sealant Military Specification MIL-S-22473, Grade H.

i. Place gear (31:) on shaft (36) and secure with retaining ring (30).

j. Lubricate all gears with grease MIL-G-23827 (GIA).

k. Proceed with assembly of torque tube assembly (24) and housing assembly (46). Apply grease MIL-G-23827 (GIA) on housing bearings. Tests to be performed during assembly are as follows:

(1) After installation of torque tube assembly (24) but prior to installation of motor and sensor assembly (12), the torque required to turn shaft (36) shall not exceed 5 inch-pounds. Use a torque wrench in the 1/4-inch square socket in the end of the shaft.

Note. If torque value exceeds 5 inchpounds, the worm gear (38) may not be properly centered on gear sector of tube assembly (24). Recheck positioning of bearing housing assemblies (39 and 35); steps a through g above. If trouble persists, remove tube assembly (24) from housing assembly (46) and inspect bearing surfaces for dents, nicks, and burrs.

(2) Prior to installation of motor and sense assembly and with the gun or an equal weight installed in the correct position, the torque required to turn shaft (36) and shall not exceed 40 inch-pounds. *I.* Proceed with installation of elevation clutch assembly (33) and elevation motor and sensor assembly (12). Coat all gear teeth with grease MIL-G-23827 (GIA).

m. When installation of elevation motor and sensor assembly (12) and gear case cover (28) is completed, and with the gun or an equal weight installed in the correct position, the torque required to turn shaft (36) shall be minimum 160 inch-pounds.

Note. If this minimum torque value is not met, clutch assembly (33) is faulty and replacement is quired.

n. Install dust seal (23) on torque tube. ment the dust seal (23) into housing assem (49) and close all gaps between housing and seal using adhesive MIL-A-46106.

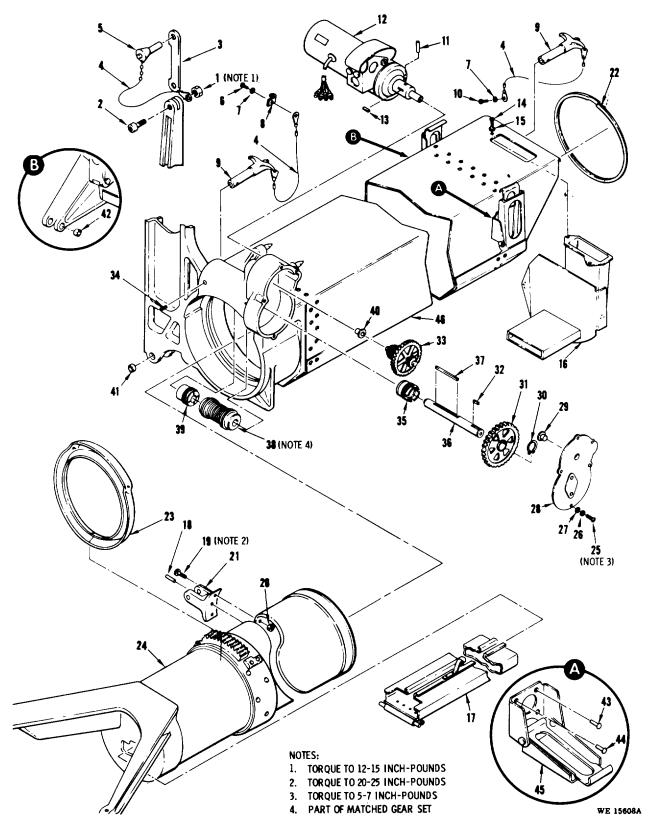


Figure 5-7. Housing and tube assembly—partially exploded view.

1-Nut	16-Adapter assembly	32-Key
2-Bolt	17-Feed chute	33-Elevation clutch assembly
3-Link	18-Locating pin	34-Set screw
4-Lanyard	19-Bolt	35-Aft bearing housing assembly
5-Release pin	20-Nut	36-Shaft
6-Screw	21-Sight drive fitting	37-Key
7-Washer	22-Retaining ring	38-Worm gear
8-Clamp	23-Dust seal	39-Forward bearing housing assembly
9-Release pin	24-Tube assembly	40-Bearing
10 -Screw	25-Screw	41-Bushing
11-Spring pin	26-Lockwasher	42-Bushing
12-Elevator motor and sensor	27-Washer	43-Rivet
assembly	28-Gear case cover	44-Rivet
13-Key	29-Bearing	45-Latch assembly
14-Screw	30-Retaining ring	46-Housing assembly
15-Washer	31-Gear	

Figure 5-7--Continued.

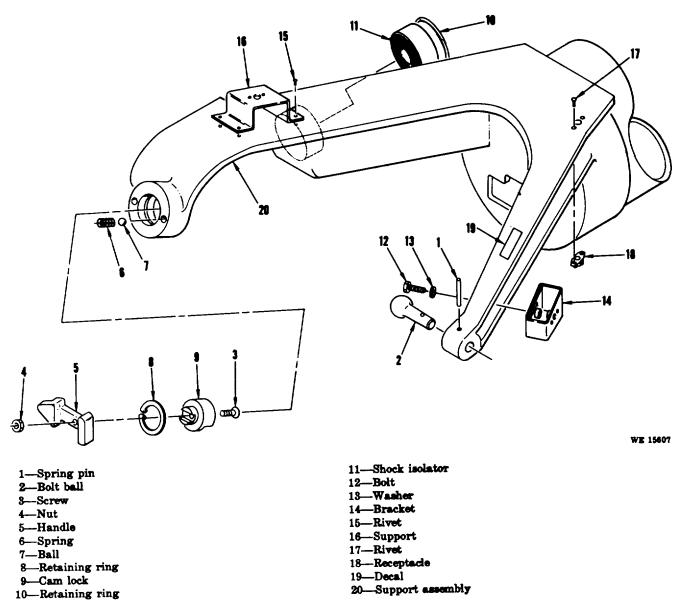
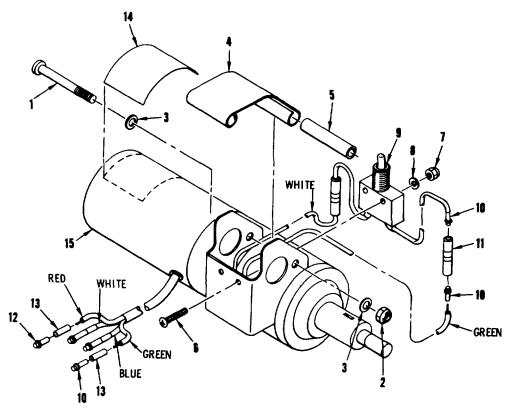


Figure 5-8. Tube assembly—partially exploded view.

5-20



Wire	From			То		
Item **	Item	Pin	Terminal	Item	Pin	Terminal
White	9	сом	*	11		10
Green	9	SSG	•	11		10
White	11		10	15		
Green	11		10	15		
White	15				TB1 - Fn	10
Green	15				TB1 - A19	10
Red	15				TB 1 - Fp	12
Blue	15				TB1 - Ep	10
			i i			

*Soldered connection **Green wire is MS25190A20; others are MS25190B16

WE61320

9-Switch

15-Motor

10-Contact, 11697831

12-Contact, 11697830

14-Identification decal

,

13-Insulating sleeve

11-Splice 11797579

1-Bolt

- 2-Nut
- 3-Washer
- 4-Switch depressor
- 5-Bushing
- 6-Screw
- 7-Nut
- 8---Washer

Figure 5-9. Motor and sensor assembly--exploded view.

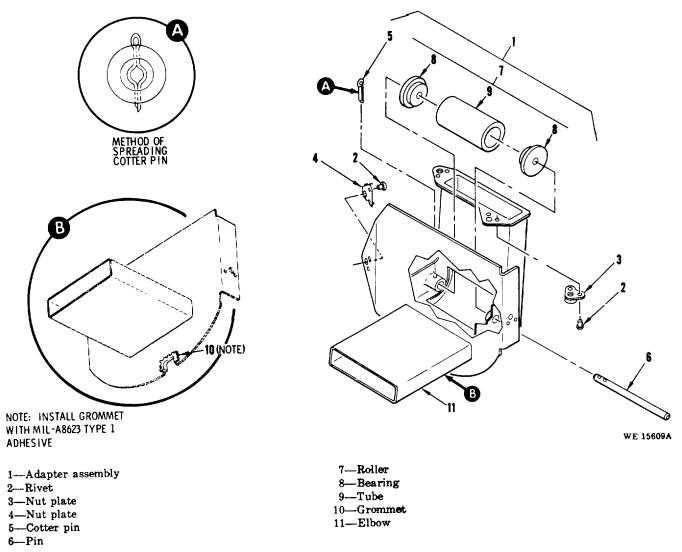
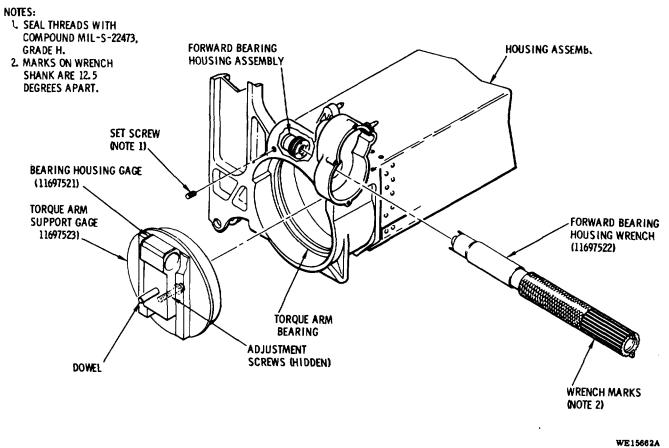


Figure 5-10. Adapter assembly-exploded view.



WE130044

Figure 5-11. Adjustment of forward bearing housing assembly.

6-1. Description

The helicopter reflex sight XM70E1 is a precision optical instrument of light construction for helicopter use. Care should be taken during all maintenance and repair work to avoid subjecting the unit to rough handling that could cause misalignment of components or breakage of optical glass.

6-2. Disassembly/Assembly

Disassemble the helicopter reflex sight XM70E1 as required for replacement of worn or damaged parts in the sequence of item numbers on figures 6-1 through 6-7. Assembly is performed in reverse of the disassembly sequence. Adjustments that are to be accomplished during assembly are outlined in paragraph 6-4.

6-3. Cleaning, Inspection, and Repair

a. Cleaning. Clean external metal surfaces with a cloth moistened with dry cleaning solvent (SD), then dry with a clean dry cloth. Do not use solvent on the beamsplitter or the projector lens. Clean these optical surfaces only with lens tissue.

b. Inspection. Check for damage, distortion, and looseness between fitted components. Check for cracked or broken optical elements.

c. Repair. Repairs are limited to replacement of worn or damaged parts and adjustment of the assembly for proper functioning.

6-4. Assembly Adjustment Requirements

During assembly of the sight, the following adjustments will be made.

a. Assemble projector lenses as shown in figure 6-8. Install and adjust reticle as follows:

(1) Install reticle lens in retainer body so that etched pattern will face the objective lens when retainer body is installed in projector. Install preformed packing, flat washer, and retainer nut in. retainer body to form a complete assembly.

(2) Install reticle retainer assembly in projector

housing.

(3) Mount the projector housing in a suitable fixture and illuminate the reticle using an auxiliary light or an unattached lamp housing.

(4) Adjust infinity focus of reticle for zero (0) ± 0.2 diopters using a dioptometer, Keuffel and Esser No. 182350 or equivalent.

Note. When reticle is properly focused, the vertical axis of the reticle pattern shall be perpendicular to projector axis A within 0o30' (zero degrees, 30 minutes). Loosen or remove retainer nut to reposition reticle lens as required. Use care to avoid turning retainer body in projector since focus will be affected.

(5) Apply sealing compound MIL-S-11031B to exposed threads of reticle retainer body to lock it in position and to produce an airtight seal.

b. Install rod assembly and adjust stowed position eccentric pin as shown in figure 6-8.

c. Adjust detent assembly, elevation knob assembly, and other assemblies as shown in figure 6-8

6-5. Sight Assembly Purging

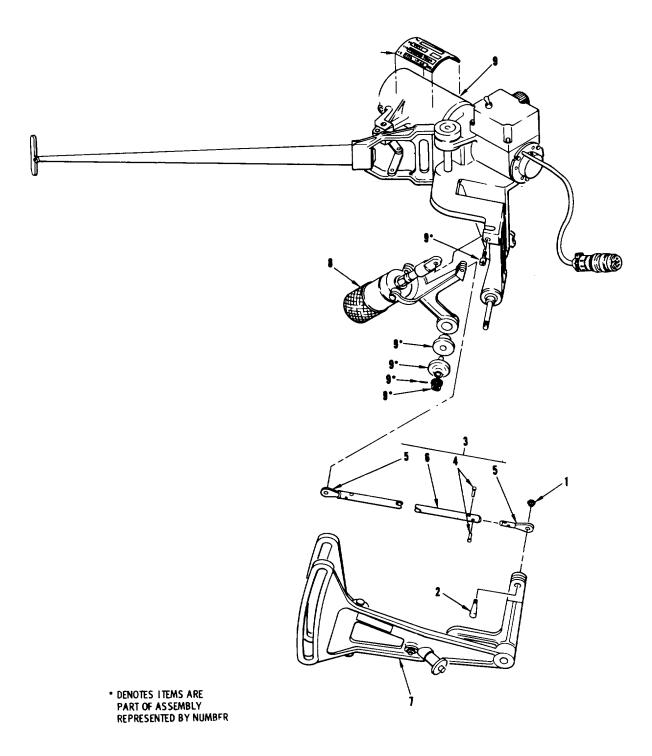
After assembly of the sight and any time the projector housing is opened and optical elements are exposed to air, the projector must be urged with dry nitrogen and resealed. Refer to TM 9-105-298-12 for procedure.

6-6. Final Inspection and Collimation

a. Set-up Instructions. See figure 6-9.

(1) Place elevation and cross leveling fixture "A" on a sturdy work table as to permit the placement of the target assembly "B" for easy viewing.

(2) Set indices at zero (0) elevation and cross leveling fixture "A" to track a plumb line. Place sight adapter "C" on plate "D" of fixture "A". Use a 5-second level to cross level sight adapter "C" before pinning.



WE 15611

Figure 6-1. Helicopter reflex sight XM70E1--partially exploded view.

1-Nut 2-Pin 3-Rod assembly 4-Rivet 5-Rod end 6-Tube 7-Support assembly 8-Elevation control assembly 9-Reticle image projector 10-Plate

(3) Place helicopter reflex sight XM70E1 in position on sight adapter "C" and lock in position by fastener knob "E".

(4) Turn elevation knob "K" until "250" on the white scale is in coincidence with knob housing index mark.

(5) Place beamsplitter arm "M" in the use position. Lock arm "F" by pin "G" at the zero (0) position.

(6) Connect a 24 VDC power supply to connector of the reticle projector, switch projector on and adjust projected image brilliance as required.

(7) The collimator target "H" shall be set so that the horizontal reticle lines are on the same plane as the horizontal projected image. Switch projector off. Plumb collimator target reticle to a plumb line. Viewing thru 3X telescope "J", superimpose telescope reticle on collimator target reticle.

b. Collimation.

(1) Switch projector on and look through 3X telescope "J" to observe projected reticle image and collimator target "H" reticle.

(2) Turn sight azimuth adjustment screw near sight fastener knob "E". Observe that line of sight can be locked at any angle between 2.5 degrees right and 0.5 degree left. Lock adjustment screw at zero angle.

c. Elevation Excursion Range.

(1) From zero (0) line of sight position, turn elevation knob "K" to the extreme range position.

Note. Depress range detent plunger when turning elevation knob into yellow marked range.

(2) Set arm "F" to 30 degrees depressed position and lock with pin "G".

(3) Return to zero (0) line of sight then move arm "F" to 15 degrees elevation position and lock with pin "G".

(4) Return to zero (0) line of sight.

d. Elevation Accuracy

(1) Make angular input changes to check the sight at various points by using arm "F".

(2) After each input change, return the projected reticle image on the beamsplitter back to zero (O) line of sight position using the elevation and cross-leveling

fixture "A" then take a reading.

(3) Angular changes in line of sight between 213 mils depression and 177 mils elevation shall be within 2.0 mils. Angular changes between 213 mils depression and 427 mils depression shall be within 4.0 mils.

e. Plumb Travel

(1) Make angular input changes as described in c (1) and d. (2) above.

(2) The projected reticle vertical line shall track a plumb line within 2.0 mils from 177 mils elevation to 213 mils depression and within 4.0 mils from 213 mils depression to 426 mils depression.

f. Superelevation Accuracy. Super elevation input changes shall be made using the elevation knob "K" then brought back to zero (0) line of sight using the elevation and cross leveling fixture "A". Superelevation inputs and tolerances are as follows:

Range (Meters)	Sight Change (Mils)	Tolerance (Mils-)
250 (White)		
500 (White)	12.01	1.5
750 (White)	18.24	2.0
1000 (White)	28.08	2.0
250 (Yellow)	-8.55	2.0
500 (Yellow)	34.90	3.0
750 (Yellow)	71.18	3.0
1000 (Yellow)	111.37	3.0
1250 (Yellow)	156.10	3.0
1500 (Yellow)	212.32	2.5

g. Height Adjustment

(1) Place mirror assembly "L" in position shown and clamp to beam-splitter arm "M".

(2) While sighting through 3X telescope "J", adjust the mirror to superimpose the telescope reticle image on its reflected image. Azimuth adjustments are made by loosening two thumb screws "P" and turning the end piece. Elevation adjustments are made using the three set screws at the lower end of the mirror assembly.

(3) Loosen height adjustment knob "R" and slip the sight up and down through its full excursion.

Caution: Ease the sight into position for check. Do not allow it to drop.

(4) The reflected reticle, viewed through 3X

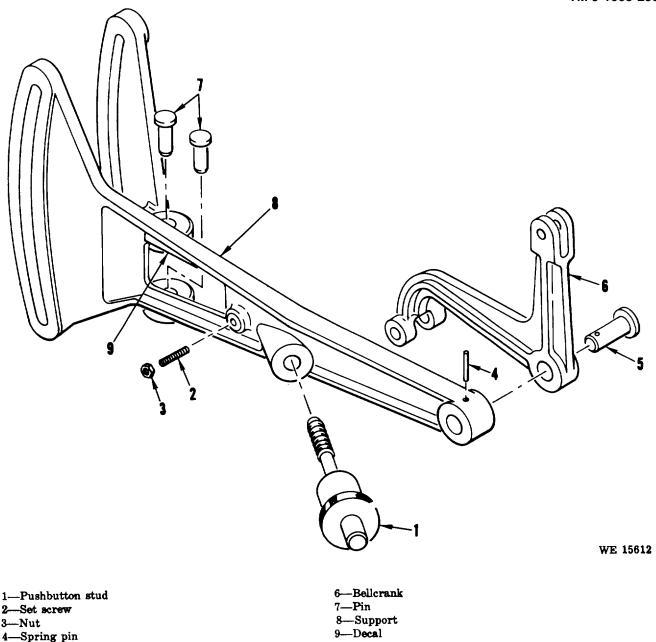




Figure 6-2. Support assembly—exploded view.

telescope "J" shall not deviate from the telescope more than 2.0 mils through the full excursion of the sight.

h. Parallax

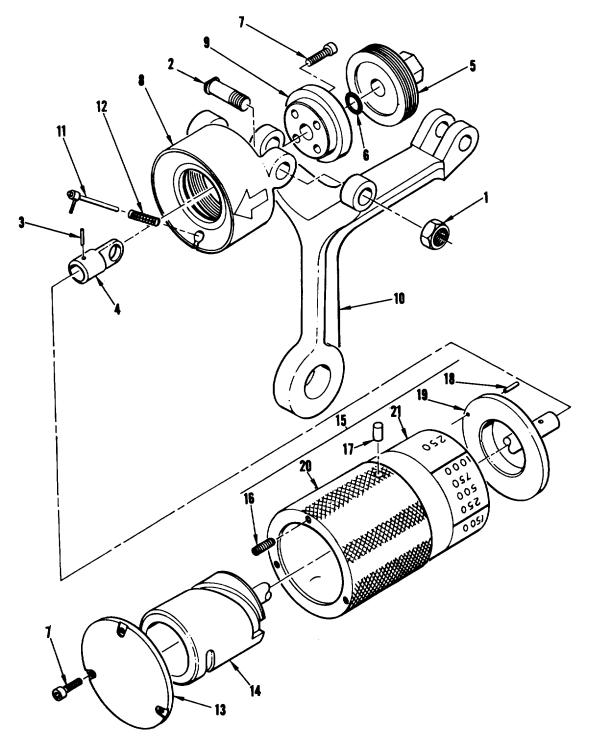
(1) Set collimator target "H" at a range between 500 and 1000 meters.

- (2) Switch on sight reticle image projector.
- (3) Remove 3X telescope "J" and place eye 8 to

10 inches from the center of the beamsplitter.

(4) Move head from side to side and up and down. Parallax between projected reticle and optical axis shall not exceed 2.0 mils over full aperture of the beamsplitter.

- i. Final Clean-up
 - (1) Disconnect 24 VDC power from the sight.
 - (2) Remove sight assembly from the test fixture.



WE 15613A

Figure 6-3. Elevation control assembly—exploded view.

1-Nut 2-Bolt 3-Spring pin 4-Coupling 5-Retainer 6-Packing 7-Screw 8-Housing 9-Bearing 10-Link 11-Plunger 12-Spring 13-Cap 14-Cam 15-Knob assembly 16-Set screw 17-Pin 18-Spring pin 19-Detent 20-Knob 21-Plate

Figure 6-3. -Continued.

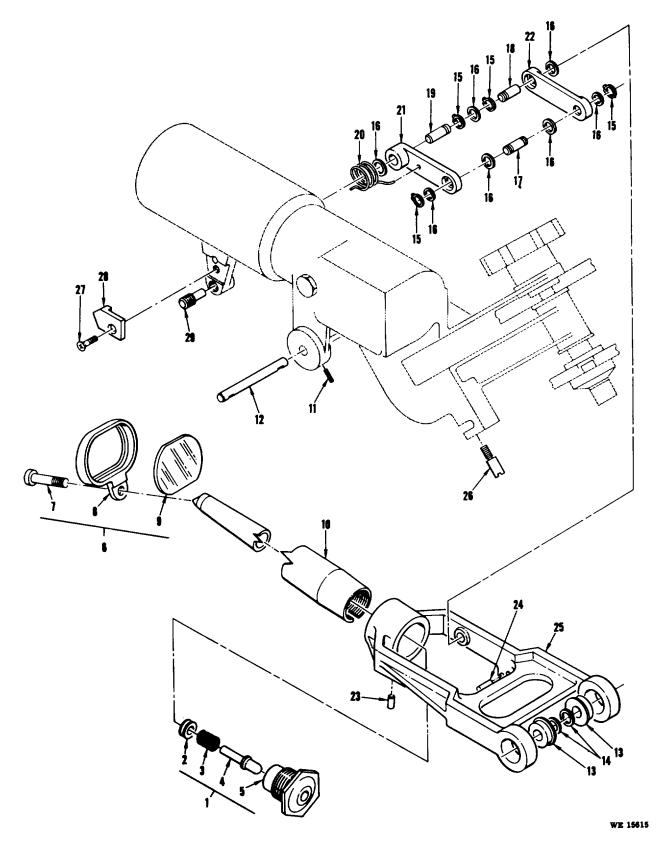
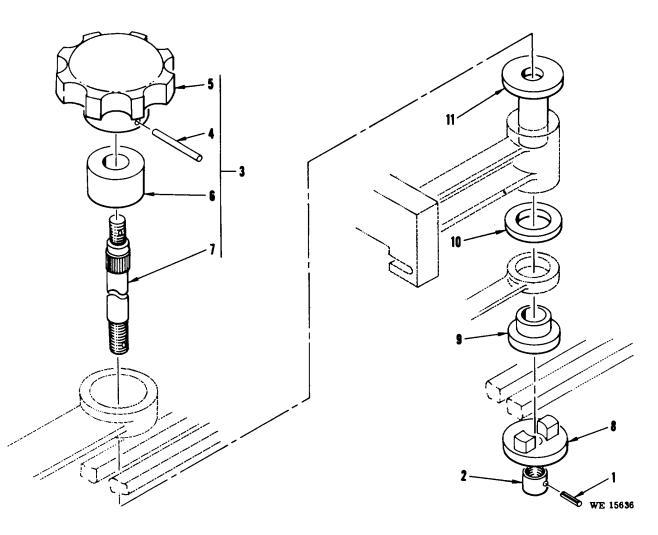


Figure 6-4. Reticle projector assembly—partially exploded view.

1-Detent assembly 2-Spring retainer 3-Spring 4-Plunger 5-Body 6-Beam splitter assembly 7-Screw 8-Retainer 9-Optical glass 10-Arm 11-Set screw 12-Shaft 13-Bearing 14-Washer 15-Retainer ring

16-Washer 17-Shaft 18-Shaft 20-Spring 21-Arm 22-Arm 23-Pin 24-Spring pin 25-Arm 26-Screw 27-Screw 28-Bracket 29-Eccentric pin

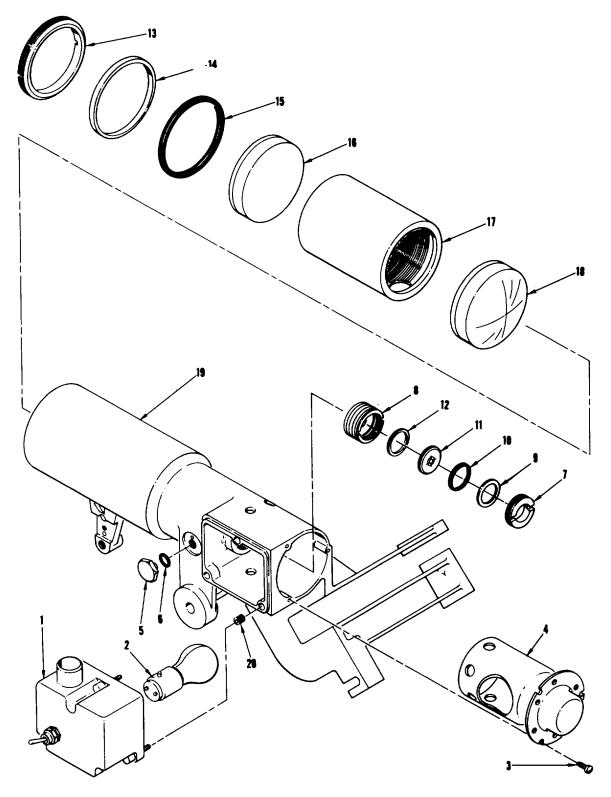
Figure 6-4—Continued.



1-Spring pin 2-Nut 3-Knob assembly 4-Spring pin 5-Knob 6-Spacer

7-Shaft 8-Tab washer 9-Bushing 10-Washer 11-Bushing

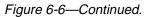


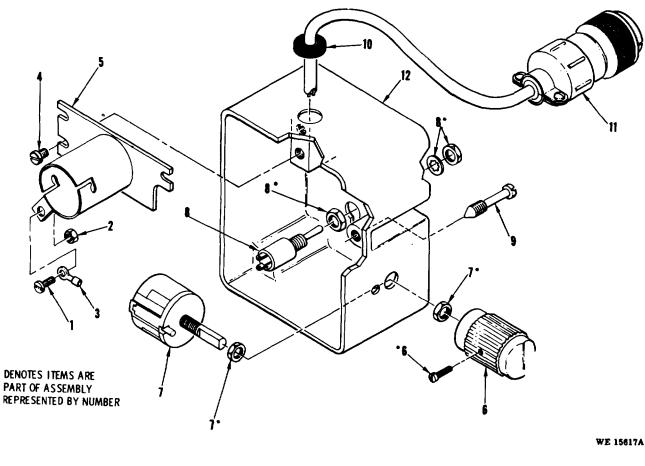


WE 15614A

Figure 6-6. Main projector housing—exploded view.

1-Lamp housing assembly 2-Lamp 3-Screw 4-Light shield 5-Plug 6-Packing 7-Retainer 8-Nut 9-Washer 10-Packing 11-Reticle 12-Nonmetallic washer 13-Retainer 14-Washer 15-Gasket 16-Lens 17-Spacer 18-Lens 19-Housing 20-Inser





1-Screw 2-Nut 3-Terminal 4-Screw 5-Lamp holder 6-Knob 7-Rheostat 8-Switch 9-Screw 10-Grommet 11-Connector 12-Housing

Figure 6-7. Lamp housing assembly—exploded view.

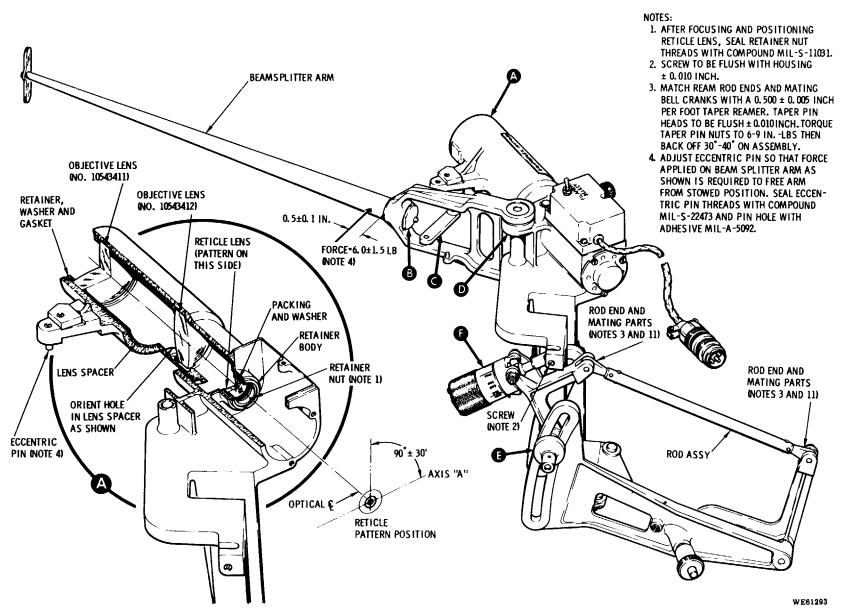
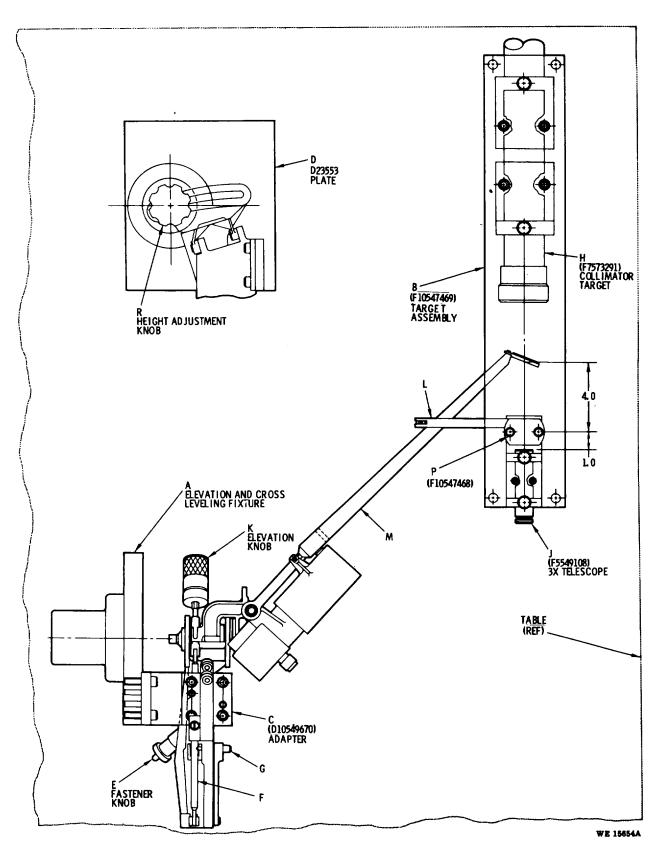
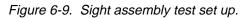


Figure 6-8 (1). Sight assembly adjustments (sheet 1 of 2).





A-Elevation and cross leveling fixture B-Target assembly C-Sight adapter D-Plate (part of fixture A) E-Sight pin F-Arm G-Pin H-Collimator target J-3X telescope K-Elevation knob L-Mirror assembly M-Beamsplitter arm P-Thumb screws R-Knob S-Surface T-Support bracket

Figure 6-9—Continued.

7-1. General

Each component is inspected during repair as specified in the chapter devoted to that component. Additional inspection of the complete system should be performed as follows:

a. Check the subsystem in accordance with preventive maintenance tables in TM 9-1005-298-12.

b. Lubricate the subsystem in accordance with lubrication chart LO 9-1005-298-12.

c. If the subsystem can be installed on an aircraft, perform operational checks in accordance with

procedures in TM 9-1005-298-12.

d. Check that identification markings correspond with those entered in equipment log (TM 38-750).

e. Check that paint and decals meet standards prescribed in TM 9-213.

f. After compliance with the instructions in this publication the armament subsystem will have been restored to a completely serviceable condition for "RETURN TO USER" or for "RETURN TO STOCK" tagging as appropriate.

APPENDIX A

REFERENCES

1. Publication Indexes

Consult each new issue of the following for the latest changes or revisions to publications listed in this appendix or fo		
new publications on the materiel covered in this manual.		
Index of Administrative Publications	DA Pam 310-1	
Index of Blank Forms	DA Pam 310-2	
Index of Doctrinal, Training, and Organizational Publications	DA Pam 310-	
Index of Supply Catalogs and Supply Manuals (Excluding types 7, 8,		
and 9)	DA Pam 31	
Index of Technical Manuals, Technical Bulletins, Supply Manuals (types		
7, 8, and 9), Supply Bulletins, and lubrication Orders	DA Pam 10-4	

2. Forms

In addition to the forms required by the Department of the Army Equipment Record System (TM B8-750), the following		
forms pertain to this materiel:		
Materiel Inspection Tag	DA Form 9-1	
Recommended Changes to DA Publications	DA Form 202	
Report of Packaging and Handling Deficiencies	DD Form 6	
Request for Issue or Turn-In	DA Form 2765 and	
	DA Form 2765-1	
Requirements for Army Aviation Publications	DA Form 12-31	
Requisition for Initial Distribution of Publications and Blank Forms	DA Form 12-1	
Requisition for Publications and Blank Forms	DA Form 17	

3. Other Publications

The following publications contain information pertinent to this materiel and associate	ed equipment.
a. Ammunition	
Ammunition and Explosives	SC-1340-9-IL
Ammunition, General	TM 9-1900
Ammunition Service in the Theater of Operations	
Care, Handling, Preservation, and Destruction of Ammunition	TM 9-1300-206
Distribution of Ammunition for Training	
Malfunctions Involving Ammunition and Explosives (reports control	
symbol AMC-132)	AR 700-1300
Regulations for Firing Ammunitions for Training, Target Practice, and	
Combat (AFR 50-13)	AR 38543
b. Camouflage	
Camouflage, Basic Principles and Field Camouflage	FM 520
c. Decontamination	
Chemical, Biological, and Radiological (CBR) Decontamination	TM 3-220
Chemical, Biological, and Nuclear Defense	

d. Destruction to Prevent Enemy Use	
Explosives and Demolitions	FM 5-25
e. General	
Accident Reporting and Records	AR 385-40
Procedures	
Basic Cold Weather Manual	FM 31-70
f. Maintenance and Repair	
Cleaning of Ordnance Materiel	TM 9-208-1
Command Maintenance Management Inspections	AR 750-8
Direct Support General Support and Depot Maintenance Repair Parts	
and Special Tools List, Armament Subsystem, Helicopter, 7.62 Milli-	
meter Machine Gun: High Rate, XM27E1	TM 9-1005-298-35P
General Maintenance Procedures for Fire Control Materiel	TM 9-254
Inspection, Care and Maintenance of Antifriction Bearings	TM 9-214
Lubrication of Ordnance Materiel	TM 9-273
Lubrication Order, Armament Subsystem, Helicopter, 7.62 Millimeter	
Machine Gun: High Rate, XM27E1	LO 9-1005-298-12
Materials Used for Cleaning, Preserving, Abrading, and Cementing	
Ordnance Materiel and Related Materials Including Chemicals	IM 9-247
Operator and Organizational Maintenance Manual, Armament Subsys-	TNA 0 4005 000 40
tem, Helicopter, 7.62 Millimeter Machine Gun: High Rate, XM27E1	IM 9-1005-298-12
Operator's Manual, Welding Theory and Application	I M 9-237
Organizational, Policies, and Responsibilities for Maintenance Operation Painting Instructions for Field Use	AR 750-5
System for Painting and Finishing Fire Control Materiel	
Adhesive, Rubber Base, General Purpose	MIL A 5002R and
Adhesive-Sealant, Silicone, RTV, General-Purpose for Electrical and	Bill-A-5902B and
Mechanical Sealing	MII -A-46106
Sealing Compound, Adhesive, Curing Polysulfide Base	
Sealing, Locking and Retaining Compounds, Single Component	MII -S-22473C GR31
Sealing, Looking and Retaining Compounds, Origie Component	and MIL-S-22473C GR20
Elementary Optics and Application to Fire Control Instruments	TM 9-258
g. Operations	111 0 200
Operation and Maintenance of Army Materiel in Extreme Cold Weather	
0° to -65°F (TO 36-1-40)	TM 9-207
h. Shipment and Storage	
DSU/Installation Stock Control and Supply Procedures (Army Field	
Stock Control Procedures)	AR 711-16
Preservation, Packaging, and Packing	AR 700-15
Requisitioning, Receipt, and Issue System	AR 725-50
Standards for Overseas Shipment or Domestic Issue of Small Arms,	
Aircraft Armament, Towed Howitzers, Mortars, Recoilless Rifles,	
Rocket Launchers, and Associated Fire Control Equipment	TB 9-1000-247-35
Storage and Materiels Handling	TM 743-200-1

A-2

By Order of the Secretary of the Army:

Official:

KENNETH G. WICKHAM, Major General, United States Army, The Adjutant General.

Distribution:

To be distributed in accordance with DA Form 1231, (qty rqr block no. 129) Direct and General Support maintenance requirements for subsystem XM27F1.

☆U.S. GOVERNMENT PRINTING OFFICE : 1990 O - 262-912 (30241)

W. C. WESTMORELAND, General, United States Army,

Chief of Staff.

RECOMMENDED CHANGES TO EQUIPMENT TECHNICAL PUBLICATIONS		
7 51	SOMETHING WRONG WITH THIS PUBLICATION?	
THEN. JOT DOWN THE DOPE ABOUT IT ON THIS FORM, CAREFULLY TEAR IT OUT, FOLD IT AND DROP IT IN THE MAIL! DATE SENT		
PUBLICATION NUMBER PUBLICATION DATE PUBLICATION TITLE		
BE EXACT PIN-POINT WHERE IT IS PAGE PARA. FIGURE NO NO	IN THIS SPACE TELL WHAT IS WRONG AND WHAT SHOULD BE DONE ABOUT IT:	
PRINTED NAME, GRADE OR TITLE, AND TELE DA 1 JUL 70 2028-2	PHONE NUMBER SIGN HERE: REVIOUS EDITIONS P.SIF YOUR OUTFIT WANTS TO KNOW ABOUT YOUR	

PIN: 026420-000